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MECHANICAL PROPERTIES, INCLUDING FRACTURE
TOUGHNESS AND FATIGUE, AND RESISTANCE TO
STRESS-CORROSION CRACKING OF STRESS-
RELIEVED STRETCHED ALUMINUM ALLOY EXTRUSIONS

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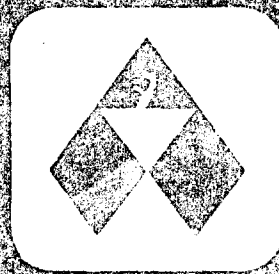
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ABSTRACT

The tensile and compressive, shear, bearing, fatigue and fracture-toughness properties of a total of 153 samples of 2014, 2024, 6061, 7075, 7079 and 7178 aluminum alloy extrusions in the TX51X and "heat-treated-by-user" tempers have been determined. The extrusions ranged in thickness from 0.050 to 6.500 in. Ratios among the tensile, compressive, shear and bearing properties have also been computed.

Stress-corrosion tests of 43 samples of extrusions have been made.

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FIFTH QUARTERLY REPORT

MECHANICAL PROPERTIES, INCLUDING FRACTURE TOUGHNESS AND FATIGUE, AND RESISTANCE TO STRESS-CORROSION CRACKING OF STRESS-RELIEVED STRETCHED ALUMINUM ALLOY EXTRUSIONS

I. Introduction.

The tests being made under this contract are for use in establishing design mechanical properties in MIL-HDBK-5A, including stress-strain and tangent-modulus curves, for 2014, 2024, 6061, 7075, 7079 and 7178 aluminum alloy extrusions in the TX51X tempers. For comparison, a limited number of similar tests are being made of extrusions in the "heat-treated-by-user" temper. Also, some fracture-toughness, axial-stress fatigue and stress-corrosion tests are being made.

This Fifth Quarterly Report summarizes the results of tensile, compressive, shear, bearing, fatigue, fracture-toughness and stress-corrosion tests made to date on 125 samples in the TX51X temper and on 28 samples in the "heat-treated-by-user" tempers. The samples ranged in thickness from 0.050 to 6.500 in.

II. Material.

A total of 138 samples of commercially-produced extrusions in the TX51X temper and 23 samples in the 0 temper have been received from two producers. The section thickness and identification of each sample is shown in Table I. Eighteen of the as-received samples in the 0 temper have been heat treated, or heat treated and aged, in accordance with applicable conditions in MIL-H-6088D. Five samples of 2024-0 and 7075-0

were tested in two "heat-treated-by-user" tempers, so that the total number of samples tested in those tempers is 28.

III. Procedure.

Mechanical Properties

The specimens and procedures used were generally in accordance with ASTM methods, and essentially in agreement with Federal Test Method 151a.

All tests were made in testing machines that meet ASTM and Government requirements for accuracy.

All tensile tests were made in accordance with ASTM Methods of Tension Testing of Metallic Materials (E8-66). The size and type of the tensile specimens are as shown in Fig. 1. Longitudinal and long-transverse specimens were taken from the following locations:

Thickness, in.	Location of Axis of Specimen with Respect to Thickness (T) and Width (W) of Predominant Section		
	Thickness	Width	
		≤ 1.500 in.	> 1.500 in.
< 0.500	T/2	---	---
0.500 to 1.500 incl.	T/2, D*/2	W/2, D*/2	W/4
> 1.500	T/4, D*/4	---	W/4, D*/4

* For round extrusions: D-diameter.

Also, for section thicknesses ≥ 0.500 and widths > 1.500 in., longitudinal and long-transverse specimens were taken at the T/2, W/2 location. For round sections > 1.500 in. in diam, specimens were also taken at the D/2 location. For sections ≥ 2.000 in. in thickness, short-transverse specimens were taken from the T/2, W/2 location.

Whenever possible, the tensile specimens from extrusions 0.499 in. or less in thickness were full-thickness sheet-type specimens (Fig. 1). The specimens from thicker shapes were 1/2 in. in diam, except where it was necessary to use subsize round specimens.

All compressive tests were made in accordance with ASTM Methods of Compression Testing of Metallic Materials (E9-67) and were made using a subpress (Fig. 3 of E9-67). The specimens from shapes less than 0.500 in. in thickness were full-thickness (Fig. 2, top). These specimens were laterally supported by a Montgomery-Templin Fixture (Fig. 4a of E9-67). The specimens from thicker shapes were cylindrical (Fig. 2, center). The compressive specimens were taken from the same locations as the tensile specimens.

Tensile and compressive yield stresses of each sample of extrusion were determined from load-strain diagrams obtained autographically.

Tests to determine the ultimate shear stress were made using specimens shown in Fig. 2 (bottom). Whenever possible, these specimens were taken from the same locations as the tensile specimens, except that tests of short-transverse specimens were made only on shapes 3 in., or more, in thickness. The tests were made with an Amsler double-shear tool in which the center 1-in. length was sheared from the 3-in. long specimen, the end thirds being supported throughout the length. In tests of longitudinal and long-transverse specimens, the loads were applied in the direction normal to the major surface of the

shape from which the specimens were taken; in tests of short-transverse specimens the loads were applied in the direction of extrusion, parallel to the major surface of the shape.

Bearing tests were made in accordance with ASTM Method E238-64T using longitudinal and, where possible, long-transverse specimens, of the types shown in Fig. 3. Flatwise and edgewise specimens were tested from shapes of suitable size. Edgewise specimens from shapes less than 1-1/2 in. in thickness, however, were 1 in. wide (Type A, Fig. 3). The bearing ultimate stresses and yield stresses were determined at edge distances of 1.5 and 2.0 times the pin diameter. The yield stress was determined as the stress at a permanent deformation of 2 per cent of the pin diameter, as indicated on autographic load-deformation diagrams. Before making these tests, the test fixtures and specimens were cleaned ultrasonically in suitable nontoxic solvent (Toson 3, Grannini Controls Corp.).

Certain lots were chosen for tensile and compressive modulus and stress-strain tests, fatigue and fracture-toughness tests. Samples from which both longitudinal and long-transverse specimens could be obtained were selected for these tests. In a few instances, however, the geometry of the shapes in certain thickness ranges permitted only longitudinal tests.

The tensile and compressive specimens used for modulus and stress-strain tests are shown in Figs. 4 and 5, respectively. In all modulus tests of longitudinal tensile specimens, and a few long-transverse specimens, strains were measured over a 6-in. gage length with an Amsler-Martens mirror-type extensometer (probably ASTM Class A). In most of the tests of long-transverse

tensile specimens it was necessary to use smaller specimens and measure strains over a 4 or 2-in. gage length with the Amsler-Martens mirror-type extensometer (ASTM Class B-1) or 1-in. gage length with the Tuckerman optical strain gage (ASTM Class A). In tests to determine modulus, the 6- or 4-in. gage length was used and the specimens were stressed up to about the proportional limit; then after removal of the load and starting again at zero stress and strain, strains were measured with the same instrument over a 2-in. gage length until the yield stress was exceeded. When strains were measured over a 2- or 1-in. gage length, tests were continued beyond the proportional limit to obtain the yield stress. In some tests of each alloy and temper, strains were measured beyond the yield stress to the ultimate stress with a 2-in. dial gage (each division = 0.001 in.) or scale and dividers to obtain complete tensile stress-strain curves. In all compressive modulus and stress-strain tests, the Tuckerman optical strain gage was used over a 2 or 1-in. gage length (ASTM Class A). For determination of each modulus value, the data were examined by the strain-deviation procedure in ASTM Method E111-61. Based on the various tests, representative typical and minimum stress-strain and tangent-modulus curves will be developed in accordance with the procedures as outlined in Section 3.2.3, 3.2.5 and 3.2.6 of Technical Report AFML-TR-66-386.*

* D. P. Moon and W. S. Hyler, "MIL-HDBK-5 Guidelines for Presentation of Data", Technical Report AFML-TR-66-386, February, 1967.

Axial-stress fatigue tests were made using three longitudinal and three long-transverse specimens of the type shown in Fig. 6. They were tested at three stress levels ($R=0.0$) in Krouse fatigue machines operating at 2400 rpm.

Fracture-toughness tests were made in accordance with the methods described in ASTM STP 411* on fatigue-cracked single-edge-notched tensile specimens from the longitudinal and long-transverse directions. The types of specimens are shown in Fig. 7; the proportions of these specimens are the same as those of specimens used by NASA, Lewis Research Center. The fracture parameters were calculated from relationships developed from the NASA calibration.

Candidate values of the critical plane-strain stress-intensity factor, K_Q , were calculated using two values of load from the fracture-toughness tests. The first value was calculated using the load at the initial burst of unstable crack growth, as indicated by the initial significant deviation from linearity in the load-deformation curve. These values were obtained from load-deformation curves developed with SR-4 electrical-resistance strain-gage units, mounted as shown in Fig. 2 of the First Quarterly Report, dated June 15, 1966. In reporting the data, the degree of clarity of the initial deviation has been indicated by the use of the letter "P" to indicate a clear instability or pop-in, the letter "I" to

* W. F. Brown and J. E. Srawley, "Plane Strain Fracture Toughness Testing of High Strength Metals", ASTM STP 411, February, 1967.

indicate a less pronounced but yet abrupt initial deviation from linearity which is believed to be a suppressed pop-in, and "M" to indicate that the initial deviation was at the maximum or fracture load. The second value was calculated using the load at a 5 per cent secant offset, equivalent to about 2 per cent of crack extension; this was done as a result of recent recommendations of ASTM Committee E-24* that the secant-offset method be considered for establishing K_Q .

Before values of K_Q can be accepted as values of K_{Ic} , they must meet two criteria:

- (a) the plastic zone size must be small with respect to the thickness, as indicated by the limitation that the thickness of the test specimen must be equal to or greater than 2.5 times the ratio $(K_Q/\sigma_{YS})^2$, and
- (b) any deviation from linearity in the load-deformation curve prior to the load used for the K_Q calculation must primarily represent crack extension, as indicated by the limitation on the load-deformation diagram that the horizontal displacement of the load-deformation curve (from the initial slope) at a load 80 per cent of that at the 5 per cent secant-offset intercept shall not be more than $1/4$ of the displacement at the 5 per cent secant-offset intercept.

Values of the ratio $(K_Q/\sigma_{YS})^2$ needed to check the thickness criterion above (a) are reported with the data. Conformance

* "Draft Recommended Practice for Notch-Bend Fracture-Toughness Testing", ASTM Committee E24, February, 1967.

with the deviation criterion (b) is still being evaluated and will be reported later, along with the final analysis to determine which values of K_Q may be considered acceptable values of K_{Ic} .

Resistance to Stress Corrosion

Stress-corrosion tests have been, or are currently being conducted, with specimens from 31 samples in the TX51X-type temper, and 12 samples in the "heat-treated-by-user" tempers.

Resistance to stress-corrosion cracking is determined with two types of specimens; longitudinal and long-transverse 0.125-inch diam tensile specimens (Fig. 8) are taken at the center line of the section thickness in all instances. For sections 0.750 inches or more in thickness short-transverse C-ring specimens (Fig. 9) are also taken, again on the center line. The tensile specimens are stressed in frames as indicated in Fig. 10 of this report, and Fig. 7 of the First Quarterly Report, dated June 15, 1966, while the C-rings are stressed in bending as shown in Fig. 8 of the First Quarterly Report; the stresses are 75 per cent of the actual tensile yield stress.

The stress-corrosion evaluations are conducted by alternate immersion in a 3.5% (by weight) NaCl solution (Sterling Granulated Salt in New Kensington tap water). The test cycle includes total immersion of the specimens for 10 min. per hour and aeration above the solution for the remaining 50 minutes per hour. The test cycle is repeated 24 times daily for a 12-week period. The test equipment, shown in Fig. 9 of the First

Quarterly Report, consists of large stationary aluminum alloy tanks which contain the salt solution, and a mechanism for raising and lowering the specimens to provide the desired cycle.

Selected tensile specimen failures are examined to verify the cause as stress-corrosion cracking, and all tensile specimens that do not fracture during exposure are tested in tension to determine the loss in tensile strength cause by corrosion.

The criteria for classifying C-ring specimens as failures are as follows:

- (a) in susceptible alloy-temper combinations, cracking in C-rings is usually well defined and readily visible at a macroscopic level (magnifications up to 10 diameters). With such alloys and tempers visual examination is sufficient and any crack readily detected at 10X magnification is considered cause for removal from test. For such combinations metallographic examination to verify the cause of failure as stress-corrosion cracking, has therefore been limited to a few representative failures.
- (b) In alloy and temper combinations developed to provide a high degree of resistance to stress-corrosion cracking, C-ring failures, when they do occur, usually develop as very fine, short cracks which are not readily detected visually.

Consequently, all C-rings of such alloy and tempers are examined metallographically after completion of the 84-day exposure period. Any evidence of stress-corrosion cracking results in the specimen being classified as a failure.

The stress-corrosion test results will be compared with existing data for aluminum alloy extruded sections to ascertain that performance was typical of that expected for the various alloy systems.

IV. Summary.

The results of tensile, compressive, shear and bearing tests of 125 samples of extrusions in the TX51X temper are shown in Tables II through IX; the corresponding properties for 28 extrusions in the "heat-treated-by-user" tempers are shown in Table X. The tensile properties of all samples exceed the values in applicable Federal Specifications. Specified minimum tensile properties for extrusions are shown in Table XI; the values shown in this table are those published in "Standards for Aluminum Mill Products", The Aluminum Association, 1967. Some of these values, as indicated in the table, are lower than those shown presently in Federal Specifications. It is understood, however, that the values in "SAMP" will be in the next revisions of the Federal Specifications.

The ratios among the tensile, compressive and shear properties of the extrusions in the TX51X tempers are shown in Tables XII through XIX and those for the "heat-treated-by-user"

tempers are shown in Table XX. The ratios among the bearing and tensile properties of the corresponding extrusions are shown in Tables XXI through XXVIII and XXIX, respectively. The ratios among the properties at different locations with regard to width and thickness are shown in Table XXX. The ratios among bearing properties obtained using edgewise specimens to those obtained using flatwise specimens are shown in Table XXXI.

The results of the axial-stress fatigue tests are shown in Figs. 11 through 16, and those of the fracture-toughness tests for the extrusions in the TX51X tempers and "heat-treated-by-user" tempers are shown in Tables XXXII and XXXIII, respectively.

The current status of the stress-corrosion tests is shown in Table XXXIV for the extrusions in the TX51X tempers and Table XXXV for the extrusion in the "heat-treated-by-user" tempers. The tests results continue to indicate typical performances for the various materials.

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V. Tables and Figures.

TABLE II

MECHANICAL PROPERTIES OF STRESS-RELIEVED STRETCHED 2014-T6510 ALUMINUM ALLOY EXTRUSIONS
(AF33(615)-3580)

Section Thick- ness, in.	Cross- Sectional Area, in. ²	Longi- tudinal Orientation	Dyrec- tion	Tensile		Elongation in 2 in. or 4D, %	Comp. Yield Stress,* psi	Shear Ultimate Stress, psi	Bearing [†]		Yield Stress, psi ^{††}	Ultimate Stress, psi ^{††}	Yield Stress, psi ^{††}	Ultimate Stress, psi ^{††}
				Ultimate Stress, psi	Yield Stress, psi				Ultimate Stress, psi	Yield Stress, psi				
0.061	0.20	317950	7/2	L	IF	9.0	64 600	--	105 600	138 800	88 200	103 900	--	--
0.070	0.24	318017 [‡]	7/2	L	IF	9.5	62 800	--	104 500	134 900	88 500	103 800	--	--
0.073	0.16	317951	7/2	L	IF	11.5	59 900	--	--	--	--	--	--	--
0.246	0.45	318134 [‡]	7/2	L	IF	10.0	61 000	44 300	110 100	136 900	94 300	107 500	--	--
0.250	3.7	340154	7/2	L	IF	12.5	63 800	46 400	109 200	141 300	93 200	113 700	--	--
0.271	0.40	317954	7/2	L	IF	5.0 [§]	69 300	45 700	106 900	139 500	93 400	108 000	--	--
0.625	0.50	317952	7/2	L	IF	6.2	63 400	--	--	--	--	--	--	--
0.625	0.55	340231	7/2	L	IF	14.3	60 300	44 700	--	--	--	--	--	--
0.750	1.4	317954	7/2	L	IF	10.7	68 000	42 100	--	--	--	--	--	--
1.657	2.2	318046	D/4	L	IF	11.5	73 100	42 800	110 000	142 800	93 600	106 200	--	--
			D/2	L	IF	7.8	68 400	41 500	105 700	135 800	88 000	103 100	--	--
			D/2	L	IF	10.5	67 200	41 500	104 800	132 500	85 700	99 700	--	--
			D/2	L	IF	7.8	68 200	--	--	--	--	--	--	--

* T - Thickness; V - Width, D - Diameter
† L - Longitudinal; IF - Long-Transverse;
‡ Offset equals 0.2 per cent
§ Producer B; all others from Producer A

** Specimens and Pictures cleaned ultrasonically in Recon 3 solvent
†† Offset equals 2 per cent or pin diameter
‡‡ Sub-size sheet-type specimen; 1/8-in. wide; 1/2-in. gage length

DATE ID

Section Number, in.	Section Area, in.	Number	Location	Direction	Tensile		Elongation in 1 in. of 40. %	Comp. Tensile Stress, psi	Strap Tensile Stress, psi	Results							
					Stress, psi					Ultimate Stress, psi		Yield Stress, psi		Ultimate Stress, psi		Yield Stress, psi	
					Stress, psi	Stress, psi				Stress, psi	Stress, psi	Stress, psi	Stress, psi	Stress, psi	Stress, psi		
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
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0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1	55 000	21 000	15.0	45 000	---	95 000	110 000	70 000	94 000	---	---	---	---
0.075	0.70	118130	7/2	1													

- T - Thickness; W - Width
- Offset equals 0.2 per cent
- Producer B; all others from Producer A

```

** specimens and fixtures cleaned ultrasonically in Toluol solvent.
** Offset equals 2 per cent of pin diameter.
?? Average of two tests; all others, single tests.
?? Sub-size sheet-type specimens; 1/8-in. wide; 1-in. gage length.
000 Sub-size sheet-type specimens; 1/8-in. wide; 1 7/16 in. gage length.
000 Samples were in the T5511 temper.

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T-2-A IV

MECHANICAL PROPERTIES OF STRESS-RELIEVED STRETCHED 2024-T3510 ALUMINUM ALLOY EXTRUSIONS
[AP33(615)-3500]

Section Thick- ness, in.	Sample Cross- Sectional Area, in. ²	Ratier	Loca- tion*	Direc- tion*	Tensile Ultimate Stress, psi	Tensile Yield Stress, psi	Elongation in 2 in. or 4D, %	Comp. Yield Stress, psi	Shear Ultimate Stress, psi	Fatigue				Corrosion			
										Ultimate Stress, psi	Field Stress, psi	Ultimate Stress, psi	Field Stress, psi	Ultimate Stress, psi	Field Stress, psi	Ultimate Stress, psi	Field Stress, psi
										8/10-1.5 8/10-1.5	8/10-1.5 8/10-1.5	8/10-1.5 8/10-1.5	8/10-1.5 8/10-1.5	8/10-1.5 8/10-1.5	8/10-1.5 8/10-1.5	8/10-1.5 8/10-1.5	8/10-1.5 8/10-1.5
0.075	0.70	318084	T/2	L	77 400	66 300	6.0	70 000	--	110 200 141 400	98 500 118 600	--	--	--	--	--	--
0.094	0.30	318134	T/2	L	77 400	66 300	3.0***	71 300	--	--	--	--	--	--	--	--	--
0.103	0.13	317887	T/2	L	77 400	66 300	7.0	65 700	--	109 000 139 500	100 000 118 000	--	--	--	--	--	--
0.106	0.11	317888	T/2	L	77 400	66 300	6.0	71 300	--	108 800 141 300	95 700 128 400	--	--	--	--	--	--
0.190	0.77	318023	T/2	L	77 400	67 700	4.0	66 800	--	107 600 141 300	95 500 117 600	--	--	--	--	--	--
0.191	0.62	317889	T/2	L	77 400	67 700	6.5	70 400	--	110 400 144 400	101 200 117 600	--	--	--	--	--	--
				LF	75 200	70 130	8.0***	69 600	--	--	--	--	--	--	--	--	--
0.295	2.0	317890	T/2	L	77 600	70 000	8.5	75 500	--	113 100 146 600	99 400 117 400	--	--	--	--	--	--
0.298	4.2	318086	T/2	L	76 600	72 200	10.0***	73 000	--	114 000 145 800	99 900 119 400	--	--	--	--	--	--
0.375	0.66	31781	T/2	L	76 600	66 400	7.5	71 100	--	118 000 143 800	103 000 115 300	--	--	--	--	--	--
0.510	10.1	317894	T/2, W/2	L	76 600	70 000	7.5***	71 500	--	109 700 141 400	98 700 106 100	--	--	--	--	--	--
0.525	1.9	318044	T/2, W/2	L	76 700	67 400	4.5	68 400	--	107 100 141 400	98 200 107 700	--	--	--	--	--	--
				LF	71 700	66 300	9.5	70 700	--	108 200 137 600	96 000 107 400	--	--	--	--	--	--
0.550	1.9	317892	T/2, W/2	L	71 700	66 300	10.0	67 500	--	--	--	--	--	--	--	--	--
0.625	3.9	317893	T/2, W/2	L	71 000	65 200	5.0	67 000	--	104 500 135 300	91 600 106 600	--	--	--	--	--	--
0.650	5.0	317894	T/2, W/2	L	71 000	65 200	10.5	67 000	--	104 500 135 300	91 600 106 600	--	--	--	--	--	--
0.825	3.9	317895	T/2, W/2	L	71 000	65 200	8.0	67 000	--	104 500 135 300	91 600 106 600	--	--	--	--	--	--
0.950	4.6	317893	T/2, W/2	L	71 000	65 200	7.5	67 000	--	104 500 135 300	91 600 106 600	--	--	--	--	--	--
1.150	5.6	318078	T/2, W/2	L	71 000	65 200	4.1	67 000	--	104 500 135 300	91 600 106 600	--	--	--	--	--	--
1.300	3.9	317895	T/2, W/2	L	71 000	65 200	8.0	67 000	--	104 500 135 300	91 600 106 600	--	--	--	--	--	--
1.450	7.3	318045	T/2, W/2	L	71 000	65 200	8.0	67 000	--	104 500 135 300	91 600 106 600	--	--	--	--	--	--
1.705	4.8	318059	T/2, W/2	L	71 000	65 200	8.0	67 000	--	104 500 135 300	91 600 106 600	--	--	--	--	--	--
2.925	8.8	318060	T/2, W/2	L	71 000	65 200	8.0	67 000	--	104 500 135 300	91 600 106 600	--	--	--	--	--	--
2.760	29.6	318079	T/2, W/2	L	71 000	65 200	8.0	67 000	--	104 500 135 300	91 600 106 600	--	--	--	--	--	--
4.000	24.0	318085	T/2, W/2	L	71 000	65 200	8.0	67 000	--	104 500 135 300	91 600 106 600	--	--	--	--	--	--
4.500	30.7	318089	T/2, W/2	L	71 000	65 200	8.0	67 000	--	104 500 135 300	91 600 106 600	--	--	--	--	--	--

* T - Transverse; W - Width
* L - Longitudinal; LF - Long-Transverse; ST - Short-Transverse
* Offset equals 0.2 per cent.
* Producers B; all others from Producer A

** Specimens and Pictures cleaned ultrasonically in Vapor 3 solvent.
** Offset equals 2 per cent of pin diameter.
*** Bending specimen (failed before reaching yield stress (2 per cent offset)).
*** Sub-size sheet-type specimen; 1-in. wide; 1-in. gage length.
*** Sub-size sheet-type specimen; 1/2-in. wide; 1/2-in. gage length.
*** Sample was in the 70511 temper.

TABLE V

MECHANICAL PROPERTIES OF STRESS-RELIEVED STRETCHED 6061-T6510 ALUMINUM ALLOY EXTRUSIONS
[AP33(61s)-3580]

Section Thickness, in.	Sample Cross-Sectional Area, in.	Location	Direction	Tensile Properties		Elongation in 2 in. or 4D, %	Comp. Yield Stress, psi	Shear Ultimate Stress, psi	Bearing ¹		Bearing ²		Yield Stress, psi	Ultimate Stress, psi	Yield Stress, psi	Ultimate Stress, psi
				Ultimate Stress, psi	Yield Stress, psi				Ultimate Stress, psi	Yield Stress, psi	Ultimate Stress, psi	Yield Stress, psi				
0.050	0.42	318136	T/2	45 000	42 400	11.0	41 500	--	79 400	107 300	66 900	75 800	--	--	--	--
0.075	0.59	317857	T/2	45 900	39 900	13.0	40 100	--	79 400	107 300	66 900	75 800	--	--	--	--
0.090	0.27	318020	T/2	44 800	42 100	15.0++	41 300	--	77 600	103 900	66 300	75 600	--	--	--	--
0.125	0.61	317847	T/2	43 200	38 700	11.5	38 600	--	77 600	103 900	66 300	75 600	--	--	--	--
0.126	0.30	317847	T/2	43 900	39 700	15.0++	39 800	31 400	--	--	--	--	--	--	--	--
0.245	1.1	340428	T/2	43 300	41 600	17.0	40 700	33 200	77 600	103 900	66 300	75 600	--	--	--	--
0.290	0.36	317848	T/2	43 500	39 900	14.1	40 800	33 800	--	--	--	--	--	--	--	--
0.294	0.97	340422	T/2	43 400	42 400	12.0	42 800	30 800	--	--	--	--	--	--	--	--
0.310	6.3	317905	T/2	47 300	42 900	16.5	46 500	32 800	78 600	102 800	67 200	76 700	--	--	--	--
0.315	5.8	317953	T/2	47 700	43 400	17.2	42 500	31 800	79 500	101 700	66 800	75 200	--	--	--	--
0.375	8.6	317927	T/2	45 500	40 200	16.5	44 800	30 700	80 700	103 400	65 700	79 100	--	--	--	--
0.375	7.7	318083	T/2	45 900	42 000	18.5	40 100	33 600	78 000	103 100	65 300	71 000	--	--	--	--
0.918	1.7	317906	T/2, W/2	46 600	42 500	17.0	42 000	36 100	80 600	103 700	66 700	75 900	--	--	--	--
1.004	2.0	340428	T/2, W/2	45 900	41 800	22.5+++	41 800	35 400	79 800	101 100	68 400	75 300	--	--	--	--
1.240	2.7	317937	T/2, W/2	45 500	42 200	19.5	41 700	33 000	78 000	101 400	68 500	75 200	--	--	--	--
				45 600	42 300	20.0	40 500	31 600	75 200	99 400	61 800	74 900	73 100	96 600	60 600	72 700
				47 000	41 900	20.0	40 600	31 900	75 300	--	--	--	73 200	96 600	61 100	74 000
				48 100	43 000	18.5	44 800	28 700	--	--	--	--	--	--	--	--
				48 600	43 700	18.5	44 800	28 400	72 600	94 300	57 900	66 800	71 900	92 100	56 200	66 900
				49 300	37 600	18.0	40 800	26 400	--	--	--	--	--	--	--	--
1.360	4.4	317896	T/2, W/2	51 600	46 800	18.5	46 800	29 600	76 000	96 600	60 000	71 000	74 700	96 300	59 700	70 600
3.000	15.0	340226	T/2, W/2	52 000	47 500	18.5	47 800	29 100	75 100	96 300	60 000	71 000	74 700	96 000	62 100	73 400
				52 900	46 900	16.0	50 100	27 800	76 100	98 600	63 200	73 200	74 900	94 400	60 700	73 400
				52 700	43 700	15.0	48 700	27 800	74 500	95 900	61 800	74 700	72 100	94 300	60 700	73 400
				52 000	46 700	12.5	48 700	27 800	75 600	97 400	62 500	72 400	74 100	94 300	60 400	70 300
				47 100	43 800	11.0	42 500	26 700	--	--	--	--	--	--	--	--
5.500	35.2	317937	D/4	51 500	46 100	13.0	44 700	28 000	72 600	93 800	57 200	71 000	73 600	92 600	57 200	71 000
				51 600	48 000	12.5	40 700	27 000	69 300	89 500	55 500	68 200	58 500	86 600	54 700	67 500
				51 000	45 700	13.5	45 600	26 600	71 500	91 500	56 500	67 500	--	--	--	--
				54 000	37 000	12.5	39 000	26 000	--	--	--	--	--	--	--	--

TABLE VI

MECHANICAL PROPERTIES OF STRESS-RELIEVED STRUTTED 7075-T6510 ALUMINUM ALLOY KATHUSIONS
[AP33(015)-1-80]

Section Thick- ness, in.	Sample Cross- Sectional Area, in.	Number	Loca- tion*	Direc- tion†	Tensile Ultimate Stress, psi	Tensile Yield Stress, psi	Elongation in 2 in. or 4D, %	Comp. Yield Stress, psi	Shear Ultimate Stress, psi	Fatigue**							
										Plate††				Rivets			
										Ultimate Stress, psi	Yield Stress, psi	Ultimate Stress, psi	Yield Stress, psi	Ultimate Stress, psi	Yield Stress, psi	Ultimate Stress, psi	Yield Stress, psi
										σ/D=1.5	σ/D=1.5	σ/D=1.5	σ/D=1.5	σ/D=1.5	σ/D=1.5	σ/D=1.5	σ/D=1.5
0.065	0.18	317899	T/2	L	87 000	74 200	11.0	75 400	--	126 600	158 400	106 300	123 100	--	--	--	--
0.065	0.27	318031#	T/2	L	84 200	77 700	10.5	77 200	--	125 900	158 700	102 400	125 400	--	--	--	--
				LF	83 000	74 500	17.0***	77 800	--	--	--	--	--	--	--	--	--
0.080	0.18	317898	T/2	L	88 200	81 500	12.0	80 500	--	125 500	157 900	109 600	124 300	--	--	--	--
0.133	0.97	318029#	T/2	L	84 200	75 800	9.0	76 500	--	124 600	156 200	107 000	123 900	--	--	--	--
				LF	82 800	72 600	15.0***	77 500	--	--	--	--	--	--	--	--	--
0.160	0.26	318030#	T/2	L	81 500	79 400	12.0	81 900	--	126 500	156 100	108 600	129 600	--	--	--	--
0.209	1.2	340403	T/2	L	83 700	82 700	11.0	88 200	91 100	125 400	158 500	110 500	134 500	--	--	--	--
				LF	89 500	80 200	12.0***	90 500	48 400	131 800	150 900	108 600	124 500	--	--	--	--
0.260	1.2	318028#	T/2	L	81 500	77 900	11.5	78 800	48 500	126 100	158 000	104 800	123 400	--	--	--	--
0.313	0.51	317908	T/2	LF	84 100	74 800	15.0***	85 300	49 400	--	--	--	--	--	--	--	--
0.325	2.4	340437	T/2	L	86 000	82 700	12.0	77 100	46 800	122 900	156 200	103 100	119 400	123 700	151 700	103 100	126 400
				LF	86 000	77 300	13.5	81 200	45 700	--	--	--	--	--	--	--	--
0.375	2.2	317954	T/2	L	86 000	76 200	11.0	79 200	48 000	125 000	150 600	104 400	126 600	--	--	--	--
				LF	85 000	78 600	13.5	82 100	47 100	--	--	--	--	--	--	--	--
0.438	7.2	317899	T/2	L	87 600	80 300	12.0	86 400	49 900	129 100	160 200	113 000	128 500	--	--	--	--
				LF	85 100	77 700	11.5	85 100	46 100	131 000	153 100	112 700	134 200	--	--	--	--
0.463	1.9	318033#	T/2	L	83 500	76 500	12.0	78 500	45 500	121 900	155 400	99 400	119 700	--	--	--	--
				LF	78 700	70 800	10.0	78 400	45 700	--	--	--	--	--	--	--	--
0.935	7.2	340155	T/2, W/A	L	88 300	81 700	11.5	80 400	47 500	127 500	161 700	108 500	127 200	--	--	--	--
				LF	83 500	76 100	12.0	81 100	46 100	129 400	157 400	107 600	131 600	--	--	--	--
			T/2, W/2	L	85 400	79 600	12.0	77 600	47 000	125 900	157 400	107 600	124 300	--	--	--	--
1.023	1.8	318031#	T/2, W/A	L	85 200	75 300	12.0	79 400	45 600	--	--	--	--	--	--	--	--
1.108	27.1	317860	T/2, W/A	L	88 800	82 400	9.0	84 200	47 100	121 000	156 100	100 000	123 100	--	--	--	--
				LF	84 500	79 300	11.0	79 500	41 100	128 500	160 400	108 600	123 400	108 300	148 100	98 100	117 200
			T/2, W/2	L	86 500	79 300	11.0	81 000	47 900	129 000	159 400	111 600	128 600	112 000	149 900	99 700	120 200
				LF	85 100	78 000	12.0	77 100	46 700	125 500	155 600	107 400	119 400	110 800	147 100	100 600	119 700
1.500	1.8	317955	D/2	L	85 000	73 400	13.0	80 000	47 500	124 000	158 000	110 800	127 600	115 800	148 400	97 500	117 200
				LF	77 600	67 500	9.5	88 000	48 200	122 400	156 300	102 300	121 400	--	--	--	--
2.000	3.1	317861	D/A	L	93 400	86 700	11.0	86 400	49 200	127 300	164 500	108 000	126 400	--	--	--	--
			D/2	L	91 100	84 100	10.0	84 800	47 700	120 700	155 700	102 300	122 900	--	--	--	--
				LF	83 500	76 000	10.0	79 300	46 300	120 600	151 400	109 000	125 000	103 200	139 200	95 400	121 700
2.190	17.0	318137#	T/A, W/A	L	79 100	66 300	6.6	76 400	45 200	119 900	149 200	104 400	124 300	99 400	131 000	96 800	119 600
				LF	81 700	73 300	10.0	72 400	44 500	117 000	149 100	100 400	120 700	101 700	136 600	98 700	120 600
			T/2, W/2	L	77 400	68 800	9.0	73 000	43 100	120 300	152 200	103 800	121 500	101 900	135 300	94 400	115 600
				LF	70 500	61 500	5.9	72 000	42 000	--	--	--	--	--	--	--	--
2.750	8.2	340404	T/A, W/A	L	91 000	83 000	11.0	84 600	46 400	122 300	158 200	103 100	122 900	123 900	155 100	105 100	122 900
				LF	80 200	69 700	9.4	77 200	47 100	121 800	157 100	100 300	120 000	120 700	154 200	102 300	118 600
			T/2, W/2	L	89 200	81 100	11.0	81 500	47 200	--	--	--	--	--	--	--	--
				LF	78 500	68 600	8.0	74 400	40 400	--	--	--	--	--	--	--	--
3.040	13.8	318138#	T/2, W/A	L	86 700	79 100	10.0	80 700	46 200	120 100	152 000	104 500	124 300	115 200	146 100	103 400	122 900
				LF	76 000	67 300	7.0	73 500	45 000	118 500	149 300	106 500	126 300	113 900	146 100	100 600	125 000
			T/2, W/2	L	84 900	76 700	10.5	77 800	45 400	118 500	151 400	101 800	121 600	115 900	145 800	101 600	118 700
				LF	75 400	66 700	8.2	72 800	43 700	--	--	--	--	--	--	--	--
				ST	73 400	63 200	6.0	70 400	43 600	--	--	--	--	--	--	--	--
3.090	24.3	340391	T/A, W/A	L	87 600	78 200	11.5	78 000	47 000	125 100	159 400	102 900	130 700	111 600	148 000	98 000	120 000
				LF	81 200	71 500	10.0	76 400	44 000	124 600	151 800	103 600	118 600	111 700	147 200	96 400	118 600
			T/2, W/2	L	84 200	74 600	11.5	78 300	44 100	117 400	150 800	97 100	116 500	108 600	136 700	94 300	113 700
				LF	77 800	68 000	8.5	73 500	43 500	118 000	151 200	95 700	113 100	104 700	135 600	91 500	112 500
				ST	75 000	66 600	8.0	71 800	44 500	--	--	--	--	--	--	--	--

* T - Thickness; W - Width; D - Diameter
† L - Longitudinal; LF - Long-Transverse; ST - Short-Transverse
‡ Offset equals 0.2 per cent.
§ Producer B; all others from Producer A

** Specimens and Pictures cleaned ultrasonically in Tescan 3 solvent.
†† Offset equals 2 per cent of pin diameter.
‡‡ Subsize sheet-type specimen, 1/2-in. wide, 1/2-in. gage length.
§§ Subsize sheet-type specimen, 1/8-in. wide, 1-in. gage length.

TABLE VII
MECHANICAL PROPERTIES OF STRESS-RELIEVED STRETCHED 7075-T73510 ALUMINUM ALLOY EXTRUSIONS
AP33(615)-3580

Section Thick- ness, in.	Sample Cross- sectional area, sq. in.	Number	Longi- tudinal	Dires- tional	Tensile		Elongation in. 2 in. or 4D, %	Comp. Yield Stress,* psi	Shear Ultimate Stress, psi	Matrices		Bearings	
					Ultimate Stress, psi	Yield Stress,* psi				Ultimate Stress, psi	Yield Stress, psi	Ultimate Stress, psi	Yield Stress, psi
0.065	0.18	317662	T/2	L	79 400	71 100	9.0	71 300	42 500	120 600	103 200	120 500	103 200
0.065	1.2	340393	T/2	L	73 300	64 100	12.0	66 900	41 500	109 400	89 100	107 200	89 100
				L	73 200	62 500	12.0*	66 900	41 500	111 900	87 500	107 600	87 500
0.313	0.51	317992	T/2	L	76 100	66 200	11.5	69 400	44 200	109 300	88 300	104 400	88 300
0.325	2.4	340456	T/2	L	73 600	61 300	12.0	61 800	39 400	109 100	88 600	106 500	88 600
0.375	2.2	317900	T/2	L	74 700	62 700	11.5	65 800	44 300	113 400	92 300	109 400	92 300
				L	74 900	64 300	11.0	66 600	42 800	118 100	98 900	117 200	98 900
0.438	7.2	317910	T/2	L	72 600	62 300	12.0	69 800	45 100	118 800	97 600	120 000	97 600
				L	72 600	67 600	12.0	71 800	44 100	116 300	96 400	114 300	96 400
0.975	7.2	340292	T/2, N/A	L	73 200	70 300	12.5	70 600	44 600	115 100	97 200	113 600	97 200
				L	73 600	67 700	12.5	71 400	43 300	116 600	96 400	113 600	96 400
1.000	5.7	340439	T/2, N/A	L	71 700	66 900	12.0	69 800	42 900	112 200	90 900	107 300	90 900
				L	76 700	66 000	13.0	66 100	42 500	114 300	91 900	111 100	91 900
1.500	1.9	317956	T/2, N/A	L	73 500	63 400	12.0	66 100	42 500	110 200	88 100	103 900	88 100
				L	74 500	64 300	12.5	64 600	43 300	111 100	93 800	111 500	93 800
2.000	3.1	317948	D/2	L	72 700	62 500	11.0	73 800	44 700	111 400	90 700	109 400	90 700
				L	79 700	72 100	6.2	72 800	44 200	112 200	92 300	109 400	92 300
2.750	8.2	340440	T/4, N/A	L	78 800	71 100	11.5	71 400	43 600	110 200	89 300	107 800	89 300
				L	77 800	70 300	6.2	69 500	43 900	107 600	86 200	103 700	86 200
				L	70 200	60 500	12.5	65 100	41 600	109 700	88 600	105 900	88 600
				L	71 000	61 800	9.4	65 900	43 000	109 700	88 600	105 900	88 600
				L	75 000	66 300	12.5	66 900	43 000	109 600	86 300	104 400	86 300
				L	75 300	59 000	9.0	62 900	40 900	109 600	86 300	104 400	86 300
				ET	69 400	58 900	8.0	62 800	41 200	109 600	86 300	104 400	86 300

* T - Thickness; W - Width; D - Diameter
 † L - Longitudinal; LT - Long-Transverse
 ** Offset equals 0.2 per cent.
 *** Specimens and pictures cleaned ultrasonically in Tron 3 solvent.
 †† Offset equals 2 per cent of pin diameter.
 †† Subsize sheet-type specimen; 1/4-in. wide; 1-in. gage length

TABLE VIII

MECHANICAL PROPERTIES OF STRESS-RELIEVED STRETCHED 7079-76510 ALUMINUM ALLOY EXTRUSIONS
(AF33(615)-3580)

Section Thick- ness in.	Sample Cross- Sectional Area, in. ²	Loca- tion ¹	Direc- tion ²	Tensile		Comp. Yield Stress, ³ psi	Shear Ultimate Stress, ⁴ psi	Properties ⁵		Yield Stress, psi ⁶	Yield Stress, psi ⁶
				Ultimate Stress, psi	Yield Stress, ³ psi			Ultimate Stress, psi	Yield Stress, psi ⁶		
0.080	0.15	340405	7/2	L	77 600	76 600	--	125 100 156 200	104 400 119 400	--	--
0.146	1.1	340406	7/2	L	85 200	76 600	--	124 600 156 100	109 300 124 100	--	--
0.161	0.72	340252	7/2	L	84 800	82 100	--	128 000 157 400	108 800 125 900	--	--
				LT	86 400	78 400	--	124 400 156 500	106 300 125 300	--	--
0.251	0.82	340253	7/2	L	84 100	84 100	--	--	--	--	--
				LT	85 700	79 700	48 400	125 100 156 100	105 200 118 600	--	--
0.500	4.2	340424	7/2, V/4	L	81 000	79 600	47 500	--	--	--	--
				LT	85 200	75 700	45 000	118 800 150 600	97 300 113 700	--	--
				LT	79 500	78 500	44 000	118 900 151 100	97 500 118 600	--	--
				L	82 400	75 100	45 200	118 800 148 900	97 300 111 600	--	--
				LT	80 700	77 400	44 500	--	--	--	--

* T - Thickness; V - Width
 † L - Longitudinal; LT - Long-Transverse
 ‡ Offset equals 0.2 per cent.
 § Producer B; all others from Producer A

** Specimens and fixtures cleaned ultrasonically in Toca 3 solvent.
 †† Offset equals 2 per cent of pin diameter.
 ‡‡ Subsize sheet-type specimen; 1/8-in. wide; 1/2-in. gage length
 §§ Subsize sheet-type specimen; 1/4-in. wide; 1-in. gage length

TABLE IX

MECHANICAL PROPERTIES OF STRESS-RELIEVED STRETCHED 7178-T6510 ALUMINUM ALLOY EXTRUSIONS
[AF33(615)-3580]

Section Thickness in.	Sample Cross-Sectional Area ¹ in. ²	Loc. tion*	Direction ²	Tensile		Elongation in 2 in. or 4D, %	Comp. Yield Stress, ³ psi	Shear Ultimate Stress, psi	Plate ⁴		Pearlite ⁵	
				Ultimate Stress, psi	Yield Stress, ⁶ psi				Ultimate Stress, psi	Yield Stress, psi	Ultimate Stress, psi	Yield Stress, psi
0.063	0.37	317902	7/2	95 400	89 900	9.5	93 400	--	133 500	118 900	117 400	138 900
			L ⁷	94 200	87 500	10.0***	93 800	--	--	--	--	--
0.066	0.35	340426 ⁸	7/2	90 700	82 000	9.0	88 400	--	135 800	174 100	111 500	136 400
			L ⁷	90 800	80 900	12.0***	88 700	--	--	--	--	--
0.142	1.0	318016	7/2	93 300	86 900	9.5	92 300	--	131 600	166 700	114 900	137 300
			L ⁷	93 200	83 400	14.0***	92 300	--	--	--	--	--
0.154	0.42	318035 ⁸	7/2	92 200	85 800	9.5	90 800	--	138 300	172 600	120 300	138 500
0.162	0.49	317303	7/2	91 900	85 400	10.2	84 000	--	132 600	164 400	114 300	129 800
		340425 ⁸	7/2	91 500	85 200	9.5	84 700	--	130 000	161 400	113 200	126 700
			L ⁷	91 500	85 200	9.0***	84 700	--	--	--	--	--
0.180	4.6	340395	7/2	92 700	85 800	10.0	85 000	--	131 800	165 500	118 300	139 000
			L ⁷	94 400	87 500	8.0	--	--	140 700	164 600	122 300	145 600
0.261	0.60	340427 ⁸	7/2	96 700	86 200	10.0	88 700	52 200	135 100	170 900	110 600	129 400
			L ⁷	95 400	89 400	10.0	89 200	51 800	132 600	166 600	112 700	125 600
0.265	0.88	317936	7/2	95 400	89 400	10.0	--	--	--	--	--	--
0.625	6.9	317997	7/2, W/4	93 600	86 900	10.7	85 500	51 800	133 400	164 800	114 300	134 900
			L ⁷	91 100	81 500	10.2	88 200	50 000	132 400	158 200	113 800	134 200
			L ⁷	92 600	85 900	10.7	83 900	51 800	131 000	166 500	113 600	129 300
0.780	1.7	340254	7/2, W/2	99 200	80 700	11.4	87 500	50 200	137 200	171 500	110 800	133 100
			L ⁷	98 200	79 900	10.0	94 300	50 600	--	--	--	--
			L ⁷	98 900	79 900	8.0	--	--	132 300	166 700	113 900	137 700
1.200	3.9	318139 ⁸	7/2, W/4	94 900	89 300	9.5	91 000	50 200	132 300	166 700	113 900	137 700
			L ⁷	93 200	87 500	8.3	--	--	132 200	166 300	111 400	139 300
			L ⁷	93 200	87 500	7.5	88 900	50 700	132 200	166 300	111 400	139 300
1.438	6.4	317957	7/2	101 300	78 900	8.5	86 400	49 500	133 100	167 600	115 700	137 100
			L ⁷	85 400	76 900	4.7	97 200	52 500	--	--	--	--
2.180	15.5	318140 ⁸	7/4, W/4	91 100	83 300	8.0	83 700	49 100	124 400	161 700	113 600	134 300
			L ⁷	82 600	75 000	5.0	82 600	46 600	128 400	155 400	114 300	134 300
			L ⁷	88 600	73 500	9.0	80 100	48 500	122 600	159 100	110 700	129 300
			L ⁷	82 400	72 500	6.3	79 200	46 000	123 600	157 600	102 100	126 500
			ST	76 600	66 600	2.6	77 300	--	--	--	--	--

* T - Thickness; W - Width
 † L - Longitudinal; L⁷ - Long-Transverse; ST - Short-Transverse
 ‡ Offset equals 0.2 per cent.
 § Producer B; all others from Producer A

** Specimens and Figures cleared ultrasonically in Form 3 solvent.
 †† Offset equals 2 per cent of min diameter.
 ‡‡ Subsize sheet-type specimen; 1/8-in. wide; 1/2-in. gage length

MECHANICAL PROPERTIES OF EXTRUSIONS IN THE "HEAT-TREATED-BY USER" TEMPER

Alloy and Temper	Section Thickness, in.	Sample Cross-Sectional Area, in. ²	Location	Direction	Tensile			Elongation in 2 in. or 4D, %	Comp. Yield Stress, ksi	Shear Ultimate Stress, ksi	Bearing ^a							
					Ultimate Stress, ksi	Yield Stress, ksi	Direction				Plate ^b		Rivets ^c					
											Ultimate Stress, ksi	Yield Stress, ksi		Ultimate Stress, ksi	Yield Stress, ksi			
2024-T62	0.185	1.0	340240\$	T/2	L	68 500	62 800	11.5	67 900	--	109 200	141 200	94 400	112 000	--	--	--	
	0.300	5.3	318084	T/2	L	71 700	65 900	11.6*	68 700	--	112 200	144 600	97 300	115 100	--	--	--	
					L	74 700	63 400	9.5	70 100	43 100	112 500	145 500	98 600	115 800	--	--	--	
					L	74 700	63 400	12.5	72 600	41 400	110 200	141 800	93 000	110 100	--	--	--	
					L	71 700	65 100	13.0	69 100	41 300	110 200	141 800	93 000	110 100	--	--	--	
	0.499	1.4	318085	T/2	L	70 300	63 800	10.2	67 300	--	--	--	--	--	--	--		
	0.064	0.27	318368\$	T/2	L	66 000	44 500	20.5	44 800	--	103 800	123 100	75 300	86 900	--	--	--	
					L	68 100	43 500	18.6*	47 500	--	98 400	121 200	74 000	87 600	--	--	--	
					L	67 100	42 400	18.0*	43 900	38 800	100 800	125 000	70 600	86 600	--	--	--	
					L	72 400	48 200	13.5	49 400	38 000	107 500	128 800	73 800	90 200	--	--	--	
L					79 800	54 800	17.5	55 200	39 400	106 800	134 500	76 000	91 800	104 300	130 300	72 800	89 800	
L					82 100	58 500	15.5	58 700	40 900	104 300	130 300	73 200	86 900	98 300	126 300	72 700	84 700	
2024-T62	0.064	0.27	318369\$	T/2	L	66 700	54 900	7.5	54 100	--	107 500	132 200	86 900	102 500	--	--	--	
					L	66 500	55 200	11.0**	57 900	--	109 300	135 400	84 100	111 700	--	--	--	--
					L	66 800	54 600	9.0**	57 500	--	100 900	130 600	81 700	99 300	--	--	--	--
					L	68 200	57 700	12.0	62 200	38 700	102 800	134 100	82 300	100 000	--	--	--	--
					L	67 700	57 700	12.0	61 000	37 600	102 800	134 100	82 300	100 000	--	--	--	--
	0.500	0.64	340244	T/2	L	71 800	58 300	11.0	61 400	40 000	101 400	131 900	82 900	101 700	96 800	127 400	81 500	100 600
					L	71 200	58 100	10.5	59 500	38 400	95 800	125 000	80 500	99 400	96 900	120 200	82 200	95 500
					L	66 100	55 300	10.5	56 200	36 300	--	--	--	--	--	--	--	--
					L	67 800	53 800	6.5	53 700	35 500	--	--	--	--	--	--	--	--
					L	62 000	51 500	6.0	53 600	33 400	--	--	--	--	--	--	--	--

T = Thickness; W = Width
 L = Longitudinal; LT = Long-Transverse; ST = Short-Transverse
 Offset equals C.2 per cent
 Producer B; all others from Producer A
 Specimens and fixtures cleaned ultrasonically in Teson 3 solvent
 Offset equals 2 per cent of pin diameter
 Subsize sheet-type specimen; 1/8-in. wide; 1/2-in. gage length

Continued

TABLE X

TABLE I (Concluded)
MECHANICAL PROPERTIES OF EXTRUSIONS IN THE "HEAT-TREATED-BY USER" TEMPER
AF33(615)-5580

Alloy and Temper	Section Thickness, in.	Sample Cross-Sectional Area, in. ²	Loco- tion [†]	Direc- tion [†]	Tensile		Comp. Yield Stress, ‡ psi	Shear Ultimate Stress, ‡ psi	Flatwise		Edgewise	
					Ultimate Stress, ‡ psi	Yield Stress, ‡ psi			Ultimate Stress, ‡ psi	Yield Stress, ‡ psi	Ultimate Stress, ‡ psi	Yield Stress, ‡ psi
6061-T62	0.246	4.6	318090	T/2	L	48 000	43 800	35 300	80 200	104 000	69 400	80 500
	1.625	3.9	418091	T/4, W/4	L	46 100	41 500	35 200	82 400	105 200	73 800	85 400
				T/2, W/2	L	45 000	39 400	32 200	75 900	98 000	67 500	77 700
					L	45 900	41 500	32 700	76 300	96 300	69 200	76 700
7075-T62	0.063	0.34	318094§	T/2	L	80 700	70 500	74 900	118 000	152 200	96 900	118 600
	0.126	0.17	318092	T/2	L	79 200	70 000	76 700	124 000	158 500	106 600	125 700
	0.300	1.7	318096	T/2	L	81 000	81 500	85 400	124 000	158 500	106 600	125 700
	1.225	21.2	318098§	T/2, W/4	L	84 800	78 000	83 000	124 900	153 200	110 200	124 000
7075-T73	0.063	0.34	318095§	T/2	L	81 000	72 400	79 300	124 000	153 200	110 200	124 000
	0.126	0.17	318093	T/2	L	84 500	75 100	78 500	124 000	153 200	110 200	124 000
	0.300	1.7	318097	T/2	L	80 000	71 400	77 000	123 200	151 100	107 300	127 800
	1.225	21.2	318099§	T/2, W/4	L	82 700	85 100	86 600	123 200	152 800	106 500	125 000
7178-T62	0.050	0.15	340247	T/2	L	96 700	88 600	91 200	135 400	166 100	117 300	133 900
	0.051	0.20	340249	T/2	L	95 800	88 100	97 700	138 400	167 900	125 900	147 000
	0.403	3.0		T/2	L	98 200	89 900	92 800	134 100	167 800	116 500	136 400
					L	92 900	82 100	93 800	141 500	164 800	120 800	142 900

* T - Thickness; W - Width; D - Diameter
† L - Longitudinal; LT - Long-Transverse; ST - Short-Transverse
‡ Offset equals 0.2 per cent
§ Producer B; all others from Producer A
¶ Specimens and fixtures cleaned ultrasonically in Tomsol 3 solvent
†† Offset equals 2 per cent of pin diameter
‡‡ Subsize sheet-type specimens; 1/2-in. wide; 1/2-in. gage length
*** Subsize sheet-type specimens; 1/4-in. wide; 1-in. gage length

(Concluded)

TABLE I (Concluded)

TABLE XI

SPECIFIED MINIMUM VALUES* FOR ALUMINUM ALLOY EXTRUSIONS
[AF33(615)-3580]

Alloy and Temper	Thickness, in.	Area, sq. in.	Tensile			Federal Specification
			Ultimate Stress, psi	Yield Stress,† psi	Elongation 2 in. or 4D, %	
2014-T62	≤0.749	All	60 000	53 000	7	QQ-A-200/2b
-T6510	≤0.499	All	60 000	53 000	7	
	0.500-0.749	All	64 000	58 000	7	
	≥0.750	≥25	68 000	60 000	7	
2024-T3510,	≤0.249	All	57 000	42 000	12	QQ-A-200/3b
-T3511	0.250-0.749	All	60 000	44 000	12	
	0.750-1.499	All	65 000	46 000	10	
	≥1.500	≥25	70 000	52 000	10	
	≥1.500	≥25, ≥32	68 000	48 000	8	
-T42	≤0.749	All	57 000	38 000	12	
	≥1.500	≥25	57 000	38 000	10	
-T6510,	0.050-0.249	All	64 000	56 000	4	
-T6511	0.250-1.499	All	66 000	58 000	5	None
	≥1.500	≥32	66 000	58 000	5	
-T62	≤0.749	-	--	--	-	
	≥1.500	-	--	--	-	
6061-T62*,	≤0.249	All	38 000	35 000	8**	QQ-A-200/8b
-T6510	≥0.250	All	38 000	35 000	10	
7075-T62*,	≤0.249	All	78 000	70 000	7	QQ-A-200/11b
-T6510	0.250-0.499	All	81 000	73 000	7	
	0.500-2.999	All	81 000	72 000	7	
	3.000-4.499	≥20	81 000	71 000	7	
	5.000-4.499	≥20, 32	78 000	70 000	6	
	4.500-5.000	≥32	78 000	68 000	6	
-T73X,††	≤0.249	-	--	--	-	None
-T73510	0.250-0.499	-	--	--	-	
	0.500-1.499	-	--	--	-	
	1.500-2.999	-	--	--	-	
	3.000-4.499	-	--	--	-	
	4.500-5.000	-	--	--	-	
7079-T62*,	≤0.249	≥20	75 000	67 000	7	QQ-A-200/12b
-T6510	0.250-0.499	≥20	77 000	68 000	7	
	0.500-1.499	≥20	78 000	70 000	7	
7178-T62*	≤0.061	≥20	79 000**	73 000**	5	QQ-A-200/13
	0.250-1.499	≥20	82 000**	74 000**	5	
-T6510	0.062-0.249	≥20	84 000**	76 000**	5	
	0.250-1.499	≥25	87 000**	78 000**	5	
	1.500-2.499	≥25	86 000	77 000**	5	

* All values are as shown in the Aluminum Association Booklet, "Standards for Aluminum Mill Products," 1967.

† Offset equals 0.2 per cent.

* In QQ-A-200/8b, 11b, 12b and 13, values for T6 temper apply also for extrusions heat treated and aged by user (T62 temper).

** Lower than in Federal specifications.

†† "T73X" signifies T73-type temper for 7075 when heat treated and aged by user. Standard designation not yet assigned.

TABLE XII
RATIOS AMONG THE TENSILE, COMPRESSIVE AND SHEAR PROPERTIES
OF STRESS-RELIEVED STRETCHED 2014-T6510 ALUMINUM ALLOY EXTRUSIONS
[AF33(615)-3580]

Section Thickness, in.	Sample Gross- Sectional Area, in. ²	Number	Loca- tion*	TUS (UT)		TUS (ST)		TUS (UT)		TUS (ST)		CYS (L)		CYS (UT)		CYS (ST)		SH (L)		SH (UT)		SH (ST)	
				TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)	TUS (L)
0.061	0.30	317950	T/2	1.11	--	--	1.10	--	--	--	--	1.04	--	--	--	--	--	--	--	--	--	--	--
0.070	0.24	318017†	T/2	--	--	--	--	--	--	--	--	1.01	--	--	--	--	--	--	--	--	--	--	--
0.077	0.16	317951	T/2	--	--	--	--	--	--	--	--	1.01	--	--	--	--	--	--	--	--	--	--	--
0.085	0.45	318130†	T/2	--	--	--	--	--	--	--	--	1.01	--	--	--	--	--	--	--	--	--	--	--
0.090	3.7	340154	T/2	1.06	--	--	1.03	--	--	--	--	1.03	1.12	--	--	--	--	0.69	--	--	--	--	--
0.271	0.40	317994	T/2	1.06	--	--	1.01	--	--	--	--	0.99	--	--	--	--	--	0.71	--	0.70	--	--	--
0.625	0.50	317952	T/2	--	--	--	--	--	--	--	--	0.97	--	--	--	--	--	0.68	--	--	--	--	--
0.625	0.35	340291	T/2	--	--	--	--	--	--	--	--	0.95	--	--	--	--	--	0.54	--	--	--	--	--
0.730	1.4	317924	T/2	0.92	--	--	0.89	--	--	--	--	1.03	0.96	--	--	--	--	0.56	--	--	--	--	--
1.637	2.2	318046	D/4 D/2	--	--	--	--	--	--	--	--	1.03	--	--	--	--	--	0.57	--	--	--	--	--
				0.92	--	--	0.91	--	--	--	--	1.03	--	--	--	--	--	0.57	--	--	--	--	--

* T - Thickness; W - Width; D - Diameter

† Producer B; all others from Producer A

TABLE XIII

RATIOS AMONG THE TENSILE, COMPRESSIVE AND SHEAR PROPERTIES
OF STRESS-RELIEVED STRUCTURED 2024-T3510 ALUMINUM ALLOY EXTRUSIONS

[AF33(615)-3580]

Section Thickness, in.	Sample		Location	T ₁ (UT) T ₁ (L)		T ₂ (UT) T ₂ (L)		T ₃ (UT) T ₃ (L)		T ₄ (UT) T ₄ (L)		T ₅ (UT) T ₅ (L)	
	Sectional Area, in. ²	Number		T ₁ (UT) T ₁ (L)	T ₁ (ST) T ₁ (L)	T ₂ (UT) T ₂ (L)	T ₂ (ST) T ₂ (L)	T ₃ (UT) T ₃ (L)	T ₃ (ST) T ₃ (L)	T ₄ (UT) T ₄ (L)	T ₄ (ST) T ₄ (L)	T ₅ (UT) T ₅ (L)	T ₅ (ST) T ₅ (L)
0.075	0.70	318132†	T/2	1.07	—	0.91	—	0.85	1.09	—	—	—	—
0.094	0.30	318019†	T/2	—	—	—	—	1.10	—	—	—	—	—
0.101	0.35	317885	T/2	—	—	—	—	0.85	—	—	—	—	—
0.106	0.31	317904	T/2	—	—	—	—	0.81	—	—	—	—	—
0.120	0.27	318018†	T/2	—	—	—	—	0.83	—	—	—	—	—
0.151	0.62	317886	T/2	0.97	—	0.92	—	0.83	1.01	—	—	—	—
0.225	2.8	317942	T/2	0.96	—	0.90	—	0.84	0.97	—	—	0.53	—
0.258	4.2	318047	T/2	0.90	—	0.85	—	0.87	0.95	—	—	0.53	—
0.275	0.62	317943	T/2	0.90	—	0.79	—	0.76	—	—	—	—	—
0.310	10.1	317926	T/2, W/2	0.86	—	0.85	—	0.88	0.92	—	—	0.53	—
0.325	1.9	318020†	T/2, W/2	0.98	—	0.88	—	0.81	0.98	—	—	0.67	—
0.550	1.8	317846	T/2, W/2	1.01	—	0.90	—	0.83	0.93	—	—	0.50	—
0.662	3.8	317945	T/2, W/2	0.84	—	0.79	—	0.87	0.92	—	—	0.51	—
0.815	3.9	340418†	T/2, W/2	0.87	—	0.83	—	0.84	0.89	—	—	0.49	—
0.950	4.6	317944	T/2, W/2	0.91	—	0.87	—	0.89	0.91	—	—	—	—
1.150	5.6	318077	T/2, W/2	0.86	—	0.83	—	0.90	0.89	—	—	—	—
1.200	3.9	317946	T/2, W/2	0.91	—	0.86	—	0.88	0.90	—	—	0.50	—
1.450	7.3	318021†	T/2, W/2	0.84	—	0.78	—	0.91	0.86	—	—	—	—
1.705	4.8	340213	T/2, W/2	0.85	—	0.81	—	0.93	0.84	—	—	0.49	—
2.520	8.8	318133†	T/2, W/2	0.85	—	0.78	—	0.95	0.85	—	—	0.47	—
4.000	24.0	340214	T/2, W/2	0.81	—	0.75	—	0.96	0.81	—	—	0.46	—
2.760	29.6	318046	T/2, W/2	0.80	—	0.75	—	0.94	0.83	—	—	0.47	—
4.500	30.7	340388	T/2, W/2	0.76	—	0.74	—	0.93	0.81	—	—	0.48	—
			T/2, W/2	0.79	—	0.75	—	0.95	0.84	—	—	0.49	—
			T/2, W/2	0.83	—	0.82	—	0.97	0.85	—	—	0.47	—

* T - Thickness; W - Width

† Producer B; all others from Producer A

‡ Samples were in the T351 temper

TABLE XIV
RATIOS AMONG TENSILE, COMPRESSIVE AND SHEAR PROPERTIES
OF STRESS-RELIEVED STRETCHED 2024-T8510 ALUMINUM ALLOY EXTRUSIONS
[AF33(615)-3580]

Section Thickness, in.	Sample Cross- Sectional Area, in. ²	Number	Loca- tion*	T _{TS} (ST) T _{TS} (L)	T _{TS} (ST) T _{TS} (L)	T _{TS} (ST) T _{TS} (L)	T _{TS} (ST) T _{TS} (L)	T _{TS} (ST) T _{TS} (L)	T _{TS} (ST) T _{TS} (L)	T _{TS} (ST) T _{TS} (L)	T _{TS} (ST) T _{TS} (L)	T _{TS} (ST) T _{TS} (L)
0.075	0.70	318022†	T/2	1.07	1.02	1.02	1.07	1.02	1.07	1.02	1.07	1.02
0.094	0.30	318134†	T/2	---	---	---	---	---	---	---	---	---
0.101	0.33	317837	T/2	---	---	---	---	---	---	---	---	---
0.106	0.31	317833	T/2	---	---	---	---	---	---	---	---	---
0.130	0.27	318023†	T/2	---	---	---	---	---	---	---	---	---
0.151	0.82	317859	T/2	1.05	1.05	1.05	1.05	1.05	1.05	1.05	1.05	1.05
0.233	2.8	317890	T/2	0.99	1.02	1.02	1.06	1.06	1.04	1.04	0.57	0.55
0.253	4.2	318082	T/2	0.97	1.02	1.02	1.07	1.07	1.05	1.05	0.57	0.57
0.375	0.62	317891	T/2	0.97	1.00	1.00	1.06	1.06	1.05	1.05	0.57	0.56
0.510	10.1	317892	T/2, W/4	0.93	1.01	1.01	1.03	1.03	1.03	1.03	0.57	0.57
0.535	1.9	318024†	T/2, W/4	0.99	1.01	1.01	1.04	1.04	1.03	1.03	0.57	0.57
0.552	1.8	317822	T/2, W/4	0.97	1.02	1.02	1.04	1.04	1.03	1.03	0.57	0.57
0.642	5.8	317894	T/2, W/4	0.93	1.01	1.01	1.03	1.03	1.03	1.03	0.57	0.57
0.845	3.9	3180410†	T/2, W/4	0.99	1.02	1.02	1.04	1.04	1.04	1.04	0.57	0.57
0.940	4.6	317823	T/2, W/4	0.95	1.02	1.02	1.03	1.03	1.03	1.03	0.57	0.57
1.150	5.6	318078	T/2, W/4	0.97	1.02	1.02	1.04	1.04	1.04	1.04	0.57	0.57
1.200	3.2	317895	T/2, W/4	0.96	1.03	1.03	1.05	1.05	1.05	1.05	0.57	0.57
1.450	7.3	318025†	T/2, W/4	0.94	1.03	1.03	1.03	1.03	1.03	1.03	0.57	0.57
1.703	4.8	3180159	T/4, W/4	0.95	0.97	0.97	1.02	1.02	1.02	1.02	0.57	0.57
2.520	8.3	3180420†	T/2, W/4	0.97	1.03	1.03	1.04	1.04	1.04	1.04	0.57	0.57
2.760	29.6	318079	T/2, W/4	0.97	0.97	0.97	1.04	1.04	1.04	1.04	0.57	0.57
4.000	24.0	318025	T/2, W/4	0.94	0.96	0.96	1.02	1.02	1.02	1.02	0.57	0.57
4.500	30.7	318039	T/2, W/4	0.97	0.93	0.93	1.01	1.01	1.01	1.01	0.57	0.57
			T/2, W/2	0.97	0.97	0.97	1.01	1.01	1.01	1.01	0.57	0.57
				0.99	1.03	1.03	1.01	1.01	1.01	1.01	0.57	0.57

* W - Width;
† Producer B; all others from Producer A.
Sample was in the T8511 temper

TABLE XVI
RATIONS AMONG THE TENSILE, COMPRESSIVE AND SHEAR PROPERTIES
OF STRESS-RELIEVED STRETCHED 7075-T6510 ALUMINUM ALLOY EXTRUSIONS
(A72(615)-2580)

06-22-05 15-2530																					
Address, In.	Sample Cross- Sectional Area, In. 2	Number	Loca- tion*	T ₁ (psi)			T ₂ (psi)			T ₃ (psi)			T ₄ (psi)			T ₅ (psi)			T ₆ (psi)		
				T ₁ (L)	T ₁ (W)	T ₁ (D)	T ₂ (L)	T ₂ (W)	T ₂ (D)	T ₃ (L)	T ₃ (W)	T ₃ (D)	T ₄ (L)	T ₄ (W)	T ₄ (D)	T ₅ (L)	T ₅ (W)	T ₅ (D)	T ₆ (L)	T ₆ (W)	T ₆ (D)
0.055	0.18	317889	T/2	--	--	--	--	--	--	--	--	--	0.95	1.00	--	--	--	--	--	--	--
0.065	0.27	318031*	T/2	0.97	--	--	0.96	--	--	--	--	--	0.99	1.00	--	--	--	--	--	--	--
0.086	0.18	317858	T/2	--	--	--	--	--	--	--	--	--	0.98	1.01	--	--	--	--	--	--	--
0.133	0.97	318029*	T/2	0.98	--	--	0.96	--	--	--	--	--	1.01	1.03	--	--	--	--	--	--	--
0.160	0.26	318030*	T/2	--	--	--	--	--	--	--	--	--	1.04	1.10	--	--	--	--	--	--	--
0.209	1.2	340403	T/2	0.98	--	--	0.97	--	--	--	--	--	1.07	1.10	--	--	--	--	--	--	--
0.360	1.2	318028*	T/2	--	--	--	--	--	--	--	--	--	1.01	--	--	--	--	--	--	--	--
0.313	0.51	317908	T/2	0.98	--	--	0.96	--	--	--	--	--	1.03	--	--	--	--	--	--	--	--
0.325	2.4	340437	T/2	1.01	--	--	0.99	--	--	--	--	--	1.00	1.05	--	--	--	--	--	--	--
0.375	2.2	317954	T/2	0.95	--	--	0.92	--	--	--	--	--	0.98	1.01	--	--	--	--	--	--	--
0.418	7.2	317859	T/2	0.93	--	--	0.97	--	--	--	--	--	1.00	1.07	--	--	--	--	--	--	--
0.483	1.9	318032*	T/2	0.94	--	--	0.93	--	--	--	--	--	1.02	1.02	--	--	--	--	--	--	--
0.935	7.2	340155	T/2, W/4	0.95	--	--	0.93	--	--	--	--	--	0.98	0.99	--	--	--	--	--	--	--
1.023	1.8	318033*	T/2, W/4	0.95	--	--	0.95	--	--	--	--	--	0.97	1.00	--	--	--	--	--	--	--
1.188	27.1	317860	T/2, W/4	0.97	--	--	0.96	--	--	--	--	--	1.02	1.02	--	--	--	--	--	--	--
1.500	1.8	317925	T/2, W/2	0.96	--	--	0.96	--	--	--	--	--	0.99	1.03	--	--	--	--	--	--	--
2.000	3.1	317861	D/2	0.87	--	--	0.78	--	--	--	--	--	1.01	--	--	--	--	--	--	--	--
2.190	17.0	318137*	D/4	0.87	--	--	--	--	--	--	--	--	1.01	0.89	--	--	--	--	--	--	--
2.190	17.0	318137*	T/4, W/4	0.86	--	--	0.91	--	--	--	--	--	1.00	1.01	--	--	--	--	--	--	--
2.750	8.2	340404	T/2, W/2	0.88	--	--	0.84	--	--	--	--	--	0.99	1.00	--	--	--	--	--	--	--
2.750	8.2	340404	T/4, W/4	0.88	--	--	0.84	--	--	--	--	--	1.02	0.93	--	--	--	--	--	--	--
2.750	8.2	340404	T/2, W/2	0.88	--	--	0.85	--	--	--	--	--	1.02	0.93	--	--	--	--	--	--	--
3.040	13.8	318138*	T/4, W/4	0.88	--	--	0.85	--	--	--	--	--	1.02	0.93	--	--	--	--	--	--	--
3.090	24.3	340391	T/2, W/2	0.89	--	--	0.87	--	--	--	--	--	1.01	0.98	--	--	--	--	--	--	--
3.090	24.3	340391	T/4, W/4	0.93	--	--	0.91	--	--	--	--	--	1.00	0.98	--	--	--	--	--	--	--
3.090	24.3	340391	T/2, W/2	0.92	--	--	0.91	--	--	--	--	--	0.99	0.98	--	--	--	--	--	--	--

* T - Thickness; W - Width, D - Diameter
+ Producer's; all others from Producer A

TABLE XVII
RATIOS AMONG THE TENSILE, COMPRESSIVE AND SHEAR PROPERTIES
OF STRESS-RELIEVED STRETCHED 7075-T73510 ALUMINUM ALLOY EXTRUSIONS
[A733(615)-3580]

Thickness, in.	Sample Cross- Sectional Area, in. ²	Number	Loca- tion	Tensile			Compressive			Shear		
				T _{TS} (U) T _{TS} (L)	T _{TS} (U) T _{TS} (L)	T _{TS} (U) T _{TS} (L)	C _{TS} (U) C _{TS} (L)	C _{TS} (U) C _{TS} (L)	C _{TS} (U) C _{TS} (L)	S _{TS} (U) S _{TS} (L)	S _{TS} (U) S _{TS} (L)	S _{TS} (U) S _{TS} (L)
0.080	0.18	317852	T/2	--	0.97	--	1.00	1.03	--	--	0.56	--
0.209	0.51	340393	T/2	0.97	--	0.98	1.04	--	--	0.54	0.87	--
0.313	0.51	317300	T/2	--	--	--	1.05	--	--	--	0.87	--
0.525	2.4	340438	T/2	1.01	--	1.02	1.01	1.06	--	0.54	0.56	--
0.718	2.2	317500	T/2	0.97	--	0.97	1.02	1.04	--	0.57	0.53	--
0.438	1.2	317910	T/2	0.95	--	1.00	1.03	1.06	--	0.57	0.58	--
0.935	7.2	340292	T/2, W/4	0.97	--	0.96	1.00	1.02	--	0.55	0.56	--
1.000	5.1	340439	T/2, W/2	0.98	--	0.97	1.00	1.02	--	0.55	0.56	--
			T/2, W/2	0.98	--	0.94	1.00	0.99	--	0.55	0.58	--
1.500	1.8	317956	D/2	0.90	--	0.85	1.02	--	--	--	0.56	--
2.000	3.1	317948	D/4	--	--	--	1.02	--	--	--	0.56	--
			D/2	0.90	--	0.86	1.02	0.93	--	--	0.56	--
2.750	8.2	340440	T/4, W/4	0.93	--	0.90	1.01	0.95	--	0.54	0.57	--
			T/2, W/2	0.92	0.97	0.89	1.01	0.95	0.95	0.55	0.57	0.55

* T - Thickness; W - Width, D - Diameter

TABLE XVIII
RATIOS AMONG THE TENSILE, COMPRESSIVE AND SHEAR PROPERTIES
OF STRESS-RELIEVED STRETCHED 7073-T6510 ALUMINUM ALLOY EXTRUSIONS
[AF33(615)-2586]

Section Thickness, in.	Sample Gross- Sectional Area, in. ²	Number	Loca- tion*	TTS (LT)		TTS (ST)		TTS (LT)		TTS (ST)		CTS (LT)		CTS (ST)		SUS (LT)		SUS (ST)	
				TTS (LT)	TTS (L)	TTS (ST)	TTS (L)	TTS (LT)	TTS (L)	TTS (ST)	TTS (L)	CTS (LT)	CTS (L)	CTS (ST)	CTS (L)	SUS (LT)	SUS (L)	SUS (ST)	SUS (L)
0.080	0.15	340405	T/2	--	--	--	--	--	--	--	--	0.99	--	--	--	--	--	--	--
0.146	1.1	340406	T/2	1.00	--	--	0.97	--	--	--	--	0.99	1.06	--	--	--	--	--	--
0.161	0.72	340252	T/2	0.97	--	--	0.95	--	--	--	--	0.99	1.06	--	--	--	--	--	--
0.251	0.82	340253	T/2	0.95	--	--	0.92	--	--	--	--	1.01	1.01	--	--	0.56	0.55	--	--
0.500	4.2	340424†	T/2, W/4	0.96	--	--	0.94	--	--	--	--	1.00	1.04	--	--	0.54	0.53	--	--
			T/2, W/2	0.98	--	--	0.96	--	--	--	--	1.01	1.02	--	--	0.55	0.54	--	--

* T - Thickness, W - Width
† Producer B; all others F - Producer A

TABLE III
RATIOS INTO THE TENSILE, COMPRESSIVE AND SHEAR PROPERTIES
OF STRESS-RELIEVED STRETCHED 7178-16510 ALUMINUM ALLOY EXTRUSIONS
[AFSS(615)-3580]

Section Thickness, in.	Sample Cross- Sectional Area, in. ²	Number	Loca- tion*	TTS(T) TTS(T)	TTS(ST) TTS(T)	TTS(T) TTS(T)	TTS(T) TTS(T)	TTS(T) TTS(T)	TTS(T) TTS(T)	TTS(T) TTS(T)	TTS(T) TTS(T)	TTS(T) TTS(T)
				TTS(T) TTS(T)	TTS(ST) TTS(T)	TTS(T) TTS(T)	TTS(T) TTS(T)	TTS(T) TTS(T)	TTS(T) TTS(T)	TTS(T) TTS(T)	TTS(T) TTS(T)	TTS(T) TTS(T)
0.063	0.37	317902	T/2	0.99	--	0.97	--	1.01	1.04	--	--	--
0.066	0.35	340126†	T/2	1.00	--	0.92	--	1.00	1.08	--	--	--
0.142	1.0	318016	T/2	0.99	--	0.96	--	0.97	1.06	--	--	--
0.154	0.42	318035†	T/2	--	--	--	--	1.06	--	--	--	--
0.162	0.49	317303	T/2	--	--	--	--	0.98	--	--	--	--
0.168	1.1	340125†	T/2	0.95	--	0.94	--	0.95	1.02	--	--	--
0.180	1.6	340125	T/2	1.02	--	1.02	--	0.99	--	--	--	--
0.261	0.50	340127†	T/2	0.94	--	0.92	--	1.03	--	--	--	--
0.265	0.88	317395	T/2	--	--	--	--	1.00	--	--	--	--
0.625	6.9	317397	T/2, W/2	0.97	--	0.94	--	0.98	1.01	--	0.53	--
0.780	1.7	340254	T/2, W/2	0.97	--	0.86	--	1.01	1.02	--	0.53	--
1.200	3.9	318159†	T/2, W/2	0.90	--	0.89	--	1.02	0.99	--	0.53	--
1.498	6.4	317957	T/2, W/2	0.92	--	0.81	--	1.02	--	--	0.53	--
2.180	15.5	318140†	T/2, W/2	0.91	0.86	0.90	0.84	1.00	0.99	0.97	0.54	0.50
				0.93		0.92		1.01	1.00		0.53	0.50

* T - Thickness; W - Width
† Producer B; all others from Producer A

TABLE XX
RATIOS AMONG THE TENSILE, COMPRESSIVE AND SHEAR PROPERTIES
OF ALUMINUM ALLOY EXTENSIONS IN THE "EAT-TREATED-BU-USER" TEMPER
[AF33(615)-2380]

Alloy and Temper	Section Thickness, in.	Sample Cross-Sectional Area, in. ²	Number	Location*	TBS (1P)		TBS (1P)		TBS (1P)		TBS (1P)		TBS (1P)		TBS (1P)		TBS (1P)	
					TBS (1P)	TBS (1P)	TBS (1P)	TBS (1P)	TBS (1P)	TBS (1P)	TBS (1P)	TBS (1P)	TBS (1P)	TBS (1P)	TBS (1P)	TBS (1P)	TBS (1P)	TBS (1P)
2024-T62	0.300	6.3	318084	T/2	1.01	0.34	1.04	1.06	1.06	0.56	0.56	0.56	0.56	0.56	0.56	0.56	0.56	0.56
	0.499	1.4	318085	T/2	0.58	0.36	1.06	1.05	1.05	0.56	0.56	0.56	0.56	0.56	0.56	0.56	0.56	0.56
2024-T42	0.064	0.27	318088†	T/2	1.06	0.98	1.01	1.06	1.06	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52
	0.085	0.27	318086	T/2	1.02	1.02	1.01	1.06	1.06	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52
	0.450	2.1	340241†	T/2	1.00	0.97	1.01	1.06	1.06	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52
	0.500	0.64	340243	T/4, W/4	0.90	0.81	1.01	1.00	1.00	0.50	0.50	0.50	0.50	0.50	0.50	0.50	0.50	0.50
	2.562	6.4	340245	T/2, W/2	0.83	0.80	1.01	0.84	0.84	0.48	0.48	0.48	0.48	0.48	0.48	0.48	0.48	0.48
					0.79	0.80	1.01	0.84	0.84	0.48	0.48	0.48	0.48	0.48	0.48	0.48	0.48	0.48
2024-T62	0.064	0.27	318089†	T/2	1.06	1.01	0.99	1.05	1.05	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55
	0.085	0.27	318087	T/2	1.05	0.98	1.07	1.05	1.05	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55
	0.450	2.1	340242†	T/2	0.95	1.00	1.04	1.05	1.04	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55
	0.500	0.64	340244	T/2	0.95	0.98	1.02	1.05	1.01	0.56	0.56	0.56	0.56	0.56	0.56	0.56	0.56	0.56
	2.562	6.4	340246	T/4, W/4	0.93	0.95	1.04	1.05	1.00	0.56	0.56	0.56	0.56	0.56	0.56	0.56	0.56	0.56
				T/2, W/2	0.91	0.96	1.04	1.05	1.00	0.56	0.56	0.56	0.56	0.56	0.56	0.56	0.56	0.56
6061-T62	0.246	4.6	318090	T/2	0.96	0.95	1.03	1.00	1.00	0.73	0.73	0.73	0.73	0.73	0.73	0.73	0.73	0.73
	1.625	3.9	318091	T/4, W/4	0.98	0.95	1.03	1.05	1.04	0.68	0.68	0.68	0.68	0.68	0.68	0.68	0.68	0.68
7075-T62	0.063	0.34	318094†	T/2	0.95	0.95	1.06	1.09	1.09	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52
	0.126	0.17	318092	T/2	0.95	0.95	1.06	1.09	1.09	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52
	0.300	1.7	318096	T/2, W/4	0.95	0.95	1.06	1.09	1.09	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52
	1.225	21.2	318098†	T/2, W/2	0.95	0.95	1.06	1.09	1.09	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52
	2.250	4.1	318100	D/4	0.86	0.81	1.02	0.88	0.88	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52
				D/2	0.93	0.89	1.02	0.94	0.94	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52	0.52
7075-T7X	0.063	0.34	318095†	T/2	0.99	0.96	1.03	1.06	1.06	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55
	0.126	0.17	318093	T/2	0.97	0.96	1.07	1.02	1.02	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55
	0.300	1.7	318097	T/2, W/4	0.94	0.92	1.06	1.03	1.03	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55
	1.225	21.2	318099†	T/2, W/2	0.95	0.97	1.05	1.03	1.03	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55
	2.250	4.1	318101	D/4	0.93	0.89	1.02	0.94	0.94	0.57	0.57	0.57	0.57	0.57	0.57	0.57	0.57	0.57
				D/2	0.99	0.99	1.05	0.94	0.94	0.57	0.57	0.57	0.57	0.57	0.57	0.57	0.57	0.57
7075-T62	0.050	0.15	340247	T/2	0.99	0.99	1.05	1.04	1.04	0.53	0.53	0.53	0.53	0.53	0.53	0.53	0.53	0.53
	0.051	0.20	318102†	T/2	0.99	0.99	1.05	1.04	1.04	0.53	0.53	0.53	0.53	0.53	0.53	0.53	0.53	0.53
	0.403	3.0	340249	T/2	0.96	0.92	1.03	1.04	1.04	0.53	0.53	0.53	0.53	0.53	0.53	0.53	0.53	0.53

* T - Thickness; W - Width, D - Diameter
† Producer B; all others Producer A

TABLE XII
RATIOS OF BEARING TO TENSILE PROPERTIES OF STRESS-RELIEVED STRETCHED
2014-T6510 ALUMINUM ALLOY EXTENSIONS

Sample			Flatwise						Edgewise					
Section Thickness, in.	Sectional Area, in. ²	Location	Number	RTS(L) RTS(L)	RTS(L) RTS(L)	RTS(L) RTS(L)	RTS(L) RTS(L)	RTS(L) RTS(L)	RTS(L) RTS(L)	RTS(L) RTS(L)	RTS(L) RTS(L)	RTS(L) RTS(L)	RTS(L) RTS(L)	RTS(L) RTS(L)
				g/D=1.5 o/D=2.0	g/D=1.5 o/D=2.0	g/D=1.5 o/D=2.0	g/D=1.5 o/D=2.0	g/D=1.5 o/D=2.0	g/D=1.5 o/D=2.0	g/D=1.5 o/D=2.0	g/D=1.5 o/D=2.0	g/D=1.5 o/D=2.0	g/D=1.5 o/D=2.0	g/D=1.5 o/D=2.0
0.061	0.30	T/2	31795C	1.57	2.07	1.42	1.67	--	--	--	--	--	--	--
0.070	0.24	T/2	31801A*	1.61	2.08	1.43	1.68	--	--	--	--	--	--	--
0.246	0.45	T/2	318130	1.72	2.14	1.56	1.78	--	--	--	--	--	--	--
0.250	3.7	T/2	34015A	1.68	2.18	1.50	1.83	--	--	--	--	--	--	--
0.271	0.40	T/2	31799A	1.58	2.07	1.46	1.69	--	--	--	--	--	--	--
0.750	1.4	T/2	31792A	1.43	1.86	1.32	1.49	--	--	--	--	--	--	--
1.657	2.2	D/4	318046	1.46	1.87	1.33	1.56	--	--	--	--	--	--	--
		D/2		1.45	1.81	1.29	1.50	--	--	--	--	--	--	--

* T - Thickness; W - Width; D - Diameter
+ Producer B; all others from Producer A
NOTE: L - Longitudinal; LT - Long-Transverse

TABLE XIII

RATIOS OF BEARING TO TENSILE PROPERTIES OF STRESS-BEHELD STRENGTHED
2024-T3510 ALUMINUM ALLOY EXTRUSIONS

[AF53(615)-3580]

Sample Cross- Section Thick- ness, in.	Number	Loca- tion*	Flatline				Edges					
			$\frac{BTS(L)}{BTS(T)}$	$\frac{BTS(L)}{BTS(T)}$	$\frac{BTS(L)}{BTS(T)}$	$\frac{BTS(L)}{BTS(T)}$	$\frac{BTS(L)}{BTS(T)}$	$\frac{BTS(L)}{BTS(T)}$	$\frac{BTS(L)}{BTS(T)}$	$\frac{BTS(L)}{BTS(T)}$		
			$c/D=1.5$	$c/D=2.0$	$c/D=2.5$	$c/D=3.0$	$c/D=1.5$	$c/D=2.0$	$c/D=2.5$	$c/D=3.0$		
0.075	318132†	T/2	1.47	1.81	1.48	1.78	---	---	---	---	---	---
0.094	318019†	T/2	1.52	2.00	1.48	1.75	---	---	---	---	---	---
0.101	317885	T/2	1.51	1.86	1.43	1.69	---	---	---	---	---	---
0.106	317904	T/2	1.59	1.95	1.49	1.76	---	---	---	---	---	---
0.120	318018†	T/2	1.56	1.98	1.48	1.71	---	---	---	---	---	---
0.151	317886	T/2	1.46	1.77	1.44	1.68	---	---	---	---	---	---
0.255	317942	T/2	1.35	1.56	1.21	1.38	1.33	1.66	1.31	1.61	---	---
0.258	318047	T/2	1.35	1.66	1.23	1.52	1.34	1.68	1.25	1.60	---	---
0.375	317943	T/2	1.67	2.09	1.61	2.06	---	---	---	---	---	---
0.510	317926	T/2, W/2	1.33	1.73	1.36	1.66	---	---	---	---	---	---
0.525	318020†	T/2, W/2	1.63	1.90	1.53	1.81	---	---	---	---	---	---
0.550	317846	T/2, W/2	1.33	1.66	1.27	1.55	---	---	---	---	---	---
0.642	317945	T/2, W/4	1.31	1.66	1.26	1.52	1.31	1.58	1.26	1.51	---	---
0.815	340418†	T/2, W/4	1.32	1.63	1.24	1.52	---	---	---	---	---	---
0.950	317944	T/2, W/4	1.34	1.65	1.27	1.55	---	---	---	---	---	---
1.150	318077	T/2, W/4	1.32	1.63	1.27	1.56	1.27	1.60	1.27	1.53	---	---
1.200	317946	T/2, W/2	1.35	1.69	1.27	1.58	---	---	---	---	---	---
1.450	318021†	T/2, W/4	1.31	1.64	1.19	1.47	---	---	---	---	---	---
1.705	340213	T/2, W/2	1.31	1.62	1.24	1.49	---	---	---	---	---	---
1.705	340213	T/4, W/4	1.31	1.62	1.24	1.54	---	---	---	---	---	---
2.520	318133†	T/2, W/2	1.32	1.62	1.24	1.50	---	---	---	---	---	---
2.520	318133†	T/4, W/4	1.28	1.59	1.25	1.53	---	---	---	---	---	---
4.000	340214	T/2, W/2	1.29	1.62	1.25	1.55	---	---	---	---	---	---
4.000	340214	T/2, W/4	1.28	1.64	1.21	1.47	1.22	1.53	1.30	1.54	---	---
4.000	340214	T/2, W/2	1.18	1.65	1.21	1.54	---	---	---	---	---	---
29.6	318040	T/4, W/4	1.22	1.53	1.25	1.50	1.20	1.51	1.27	1.55	1.11	1.21
29.6	318040	T/2, W/2	1.26	1.61	1.30	1.59	1.18	1.50	1.16	1.46	1.53	1.50
30.7	340368	T/4, W/4	1.27	1.63	1.26	1.54	1.19	1.51	1.24	1.46	1.12	1.44
30.7	340368	T/2, W/2	1.31	1.63	1.29	1.51	1.24	1.49	1.25	1.44	1.39	1.43
30.7	340368	T/2, W/2	1.31	1.63	1.29	1.51	1.24	1.49	1.25	1.44	1.39	1.43

TABLE XIII
RATIOS OF BEARING TO TENSILE PROPERTIES OF STRESS-RELIEVED STRETCHED
2024-T8510 ALUMINUM ALLOY EXTENSIONS

[A73(615)-7580]

Section Thick- ness, In.	Sample Geo- metrical Area, In. ²	Number	Loca- tion	Platens						Rounded					
				R _{TS} (L)		R _{TS} (W)		R _{TS} (T)		R _{TS} (L)		R _{TS} (W)		R _{TS} (T)	
				0.75-1.5	1.5-2.0	2.0-2.5	2.5-3.0	3.0-3.5	3.5-4.0	0.75-1.5	1.5-2.0	2.0-2.5	2.5-3.0	3.0-3.5	3.5-4.0
0.075	0.70	318022†	T/2	1.52	1.95	1.45	1.68	1.47	1.88	1.47	1.88	1.41	1.68	1.41	1.68
0.094	0.30	318134†	T/2	1.20	1.91	1.44	1.71	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
0.101	0.35	317887	T/2	1.46	1.92	1.47	1.73	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
0.108	0.31	317888	T/2	1.53	2.04	1.44	1.78	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
0.120	0.27	318023†	T/2	1.54	2.00	1.44	1.63	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
0.151	0.82	317889	T/2	1.57	2.03	1.51	1.76	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
0.255	2.8	317890	T/2	1.46	1.89	1.40	1.66	1.47	1.88	1.47	1.88	1.41	1.68	1.41	1.68
0.268	4.2	318082	T/2	1.48	1.91	1.48	1.69	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
0.275	0.62	317891	T/2	1.52	1.95	1.47	1.64	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
0.310	10.1	317892	T/2	1.46	1.88	1.45	1.67	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
0.525	1.9	318024†	T/2, W/4	1.47	1.88	1.40	1.63	1.41	1.82	1.41	1.82	1.36	1.63	1.36	1.63
0.525	1.9	317922	T/2, W/4	1.40	1.80	1.40	1.63	1.41	1.82	1.41	1.82	1.36	1.63	1.36	1.63
0.642	5.8	317894	T/2, W/4	1.46	1.87	1.42	1.67	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
0.815	3.9	340419†	T/2, W/4	1.45	1.84	1.41	1.64	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
0.950	4.6	317893	T/2, W/4	1.41	1.87	1.43	1.72	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
1.150	5.6	318078	T/2, W/4	1.47	1.90	1.45	1.72	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
1.200	3.9	317895	T/2, W/4	1.42	1.81	1.45	1.61	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
1.550	7.3	318025†§	T/2, W/4	1.43	1.81	1.45	1.64	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
1.705	4.8	340169	T/4, W/4	1.47	1.86	1.42	1.66	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
2.530	8.8	340420†	T/4, W/4	1.48	1.90	1.48	1.65	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
2.760	29.6	318079	T/2, W/4	1.41	1.84	1.43	1.63	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
4.000	24.0	340225	T/4, W/4	1.27	1.75	1.42	1.67	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
4.500	30.7	340489	T/4, W/4	1.39	1.85	1.41	1.69	1.45	1.88	1.45	1.88	1.44	1.58	1.44	1.58
			T/2, W/4	1.41	1.85	1.43	1.71	1.44	1.88	1.44	1.88	1.43	1.69	1.43	1.69
			T/2, W/4	1.41	1.88	1.43	1.71	1.44	1.88	1.44	1.88	1.43	1.69	1.43	1.69

* T - Thickness, W - Width
† Producer B; all others from Producer A
‡ Sample was in the T851 temper
§ Bearing specimen failed before reaching yield stress (2 per cent offset).
NOTE: L - Longitudinal; W - Long-Transverse

TABLE XIV
RATIOS OF YIELDING TO TENSILE PROPERTIES OF STRESS-RELIEVED STRETCHED
6061-T6510 ALUMINUM ALLOY EXTRUSIONS
[AF33(615)-3580]

Section Thick- ness, In.	Cross- sectional Area, In. ²	Loca- tion*	Number	Plate										Extrusion																																																																																																																																																																																																																																																																							
				e/D=1.5					e/D=2.0					e/D=2.5					e/D=3.0					e/D=3.5					e/D=4.0																																																																																																																																																																																																																																																								
				$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$	$\frac{RHS(L)}{RHS(L)}$

* T - Thickness; W - Width; D - Diameter
* Producer B; all others from Producer A
NOTE: L - Longitudinal; LT - Long-Transverse

TABLE XIV
RATIOS OF READING TO TENSILE PROPERTIES OF STRESS-RELIEVED STRENGTHED
7075-T6510 ALUMINUM ALLOY EXTENSIONS
(N732(615)-7580)

Section Thick- ness, in.	Sample Cross- Sectional Area, in. ²	Number	Loca- tion*	Flange											
				RHS(L)				RHS(LT)				RHS(1/2)			
				σ/D=1.5	σ/D=2.0	σ/D=1.5	σ/D=2.0	σ/D=1.5	σ/D=2.0	σ/D=1.5	σ/D=2.0	σ/D=1.5	σ/D=2.0	σ/D=1.5	σ/D=2.0
0.065	0.18	317899	T/2	1.46	1.82	1.35	1.55	--	--	--	--	--	--	--	--
0.065	0.27	318031†	T/2	1.48	1.86	1.32	1.61	--	--	--	--	--	--	--	--
0.066	0.18	317858	T/2	1.42	1.81	1.34	1.53	--	--	--	--	--	--	--	--
0.135	0.97	318029†	T/2	1.48	1.86	1.41	1.63	--	--	--	--	--	--	--	--
0.160	0.26	318030†	T/2	1.45	1.83	1.37	1.62	--	--	--	--	--	--	--	--
0.209	1.2	340403	T/2	1.38	1.75	1.34	1.62	1.45	1.77	1.29	1.51	--	--	--	--
0.260	1.23	318028†	T/2	1.47	1.85	1.35	1.58	--	--	--	--	--	--	--	--
0.325	2.4	340437	T/2	1.43	1.82	1.33	1.54	1.44	1.76	1.33	1.64	--	--	--	--
0.375	2.2	317954	T/2	1.42	1.82	1.29	1.59	--	--	--	--	--	--	--	--
0.438	7.2	317859	T/2	1.47	1.83	1.41	1.60	1.50	1.75	1.40	1.67	--	--	--	--
0.463	1.9	318032†	T/2	1.45	1.85	1.30	1.56	--	--	--	--	--	--	--	--
0.935	7.2	340155	T/2, W/4	1.45	1.83	1.33	1.56	1.47	1.79	1.32	1.61	--	--	--	--
1.023	1.8	318033	T/2, W/2	1.46	1.82	1.33	1.56	--	--	--	--	--	--	--	--
1.188	27.1	317860	T/2, W/4	1.37	1.77	1.21	1.49	--	--	--	--	--	--	--	--
1.500	1.8	317955	T/2, W/2	1.47	1.83	1.36	1.52	1.49	1.82	1.40	1.61	1.26	1.23	1.25	1.51
2.000	3.1	317861	D/2	1.31	1.67	1.18	1.40	1.51	1.88	1.42	1.64	1.30	1.28	1.74	1.50
2.190	17.0	318137†	D/4	1.32	1.71	1.22	1.46	--	--	--	--	--	--	--	--
2.750	8.2	340404	T/2, W/2	1.44	1.81	1.38	1.64	1.43	1.78	1.37	1.64	1.23	1.21	1.56	1.57
				1.44	1.82	1.38	1.65	1.47	1.86	1.42	1.66	1.24	1.24	1.66	1.58
				1.35	1.74	1.24	1.48	--	--	--	--	1.36	1.27	--	--
				1.36	1.76	1.24	1.48	--	--	--	--	1.35	1.26	--	--
3.040	13.8	318138†	T/4, W/4	1.39	1.75	1.32	1.57	1.37	1.72	1.35	1.60	1.33	1.29	1.69	1.58
3.090	24.3	340391	T/2, W/4	1.40	1.79	1.33	1.58	--	--	--	--	1.35	1.32	--	--
			T/4, W/4	1.43	1.82	1.32	1.54	1.42	1.73	1.32	1.52	1.27	1.25	1.68	1.52
			T/2, W/2	1.39	1.81	1.30	1.56	1.40	1.80	1.28	1.52	1.29	1.26	1.61	1.51

* T - Thickness, W - Width; D - Diameter
† Producer B; all others from Producer A
NOTE: L - Longitudinal; LT - Long-Transverse

TABLE XXVI
RATIOS OF MEASURING TO RESIDUE PROPERTIES OF STRESS-RELIEVED STRETCHED
7075-T73510 ALUMINUM ALLOY EXTRUSIONS
[AF33(615)-3580]

Section Thick- ness, in.	Gross- Sec- tional Area, sq. in.	Loca- tion*	Flatwise						Longwise					
			R _{TS} (1)		R _{TS} (LT)		R _{TS} (LT)		R _{TS} (1)		R _{TS} (LT)		R _{TS} (LT)	
			L/D=1.5	W/D=2.0	L/D=1.5	W/D=2.0	L/D=1.5	W/D=2.0	L/D=1.5	W/D=2.0	L/D=1.5	W/D=2.0	L/D=1.5	W/D=2.0
0.080	0.18	T/2	1.45	1.84	1.45	1.69	1.49	1.97	1.37	1.68	--	--	--	--
0.209	1.2	T/2	1.45	1.85	1.45	1.67	1.49	1.97	1.37	1.68	--	--	--	--
0.255	2.1	T/2	1.49	1.92	1.44	1.70	1.48	1.92	1.45	1.74	--	--	--	--
0.375	2.2	T/2	1.51	1.95	1.44	1.70	--	--	--	--	--	--	--	--
0.430	7.2	T/2	1.53	1.98	1.44	1.74	1.54	1.98	1.45	1.78	--	--	--	--
0.995	7.2	T/2, W/4	1.47	1.89	1.47	1.63	1.45	1.82	1.38	1.62	--	--	--	--
1.000	5.7	T/2, W/4	1.50	1.92	1.43	1.66	1.49	1.92	1.37	1.66	--	--	--	--
		T/2, W/2	1.48	1.91	1.33	1.62	--	--	--	--	--	--	--	--
1.500	1.8	D/2	1.39	1.83	1.38	1.55	--	--	--	--	--	--	--	--
2.000	3.1	D/4	1.41	1.85	1.28	1.52	--	--	--	--	--	--	--	--
		D/2	1.44	1.85	1.31	1.56	--	--	--	--	--	--	--	--
2.750	8.2	T/4, W/4	1.43	1.87	1.33	1.57	--	--	--	--	--	--	--	--
		T/2, W/2	1.43	1.85	1.33	1.56	--	--	--	--	--	--	--	--

* T - Thickness; W - Width; D - Diameter
NOTE: L - Longitudinal; LT - Long-Transverse

TABLE XXVII
RATIOS OF MEASUREMENTS TO TENSILE PROPERTIES OF STRESS-RELIEVED STRETCHED
7079-T6510 ALUMINUM ALLOY EXTRUSIONS

[AF33(615)-3580]

Section Thick- ness, in.	Gross Sec- tional Area, in. ²	Loca- tion*	Number	Tensile						Fatigue					
				R _{TS} (L) R _{TS} (L) σ/D=1.5 σ/D=2.0		R _{TS} (LT) R _{TS} (LT) σ/D=1.5 σ/D=2.0		R _{TS} (LT) R _{TS} (LT) σ/D=1.5 σ/D=2.0		R _{TS} (LT) R _{TS} (LT) σ/D=1.5 σ/D=2.0		R _{TS} (LT) R _{TS} (LT) σ/D=1.5 σ/D=2.0		R _{TS} (LT) R _{TS} (LT) σ/D=1.5 σ/D=2.0	
				σ/D=1.5	σ/D=2.0	σ/D=1.5	σ/D=2.0	σ/D=1.5	σ/D=2.0	σ/D=1.5	σ/D=2.0	σ/D=1.5	σ/D=2.0	σ/D=1.5	σ/D=2.0
0.080	0.15		340405	T/2	1.46	1.86	1.35	1.54	--	--	--	--	--	--	--
0.146	1.1		340406	T/2	1.46	1.83	1.41	1.60	1.50	1.85	1.37	1.62	--	--	--
0.161	0.72		340252	T/2	1.44	1.81	1.34	1.55	--	--	--	--	--	--	--
0.251	0.82		340253	T/2	1.46	1.82	1.33	1.50	--	--	--	--	--	--	--
0.500	4.2		340424†	T/2, V/A T/2, V/2	1.43	1.81	1.28	1.50	1.43	1.82	1.29	1.57	--	--	--
					1.44	1.81	1.31	1.50	--	--	--	--	--	--	--

* T - Thickness; V - Width
† Producer B; all others from Producer A
NOTE: L - Longitudinal; LT - Long-Transverse

TABLE XVIII
RATIOS OF BEARING TO TENSILE PROPERTIES OF STRESS-RELIEVED STRETCHED
7178-T6510 ALUMINUM ALLOY EXTRUSIONS
(A733(615)-3580)

Section Thick- ness, in.	Sec- tional Area, in. ²	Number	Loca- tion*	Flatness				Edges			
				R _{TS} (L) R _{TS} (T)		R _{TS} (LT)		R _{TS} (L)		R _{TS} (T)	
				0/D=1.5	0/D=2.0	0/D=1.5	0/D=2.0	0/D=1.5	0/D=2.0	0/D=1.5	0/D=2.0
0.063	0.37	317902	T/2	1.40	1.77	1.31	1.55	--	--	--	--
0.066	0.35	340426†	T/2	1.50	1.92	1.36	1.66	--	--	--	--
0.142	1.0	318016	T/2	1.41	1.79	1.32	1.58	--	--	--	--
0.154	0.42	318035†	T/2	1.50	1.87	1.40	1.61	--	--	--	--
0.162	0.49	317903	T/2	1.44	1.79	1.34	1.52	--	--	--	--
0.162	1.1	340425†	T/2	1.42	1.76	1.32	1.52	--	--	--	--
0.180	4.6	340395	T/2	1.42	1.79	1.38	1.62	1.52	1.78	1.42	1.70
0.261	0.60	340427†	T/2	1.46	1.84	1.38	1.50	--	--	--	--
0.265	0.88	317996	T/2	1.39	1.75	1.26	1.40	--	--	--	--
0.685	6.9	317997	T/2, W/4	1.46	1.76	1.32	1.55	1.41	1.69	1.31	1.54
0.80	1.7	340254	T/2, W/2	1.41	1.80	1.32	1.51	--	--	--	--
1.200	3.9	318159†	T/2, W/2	1.38	1.72	1.19	1.41	--	--	--	--
1.478	6.4	317957	T/2, W/2	1.42	1.78	1.27	1.59	--	--	--	--
2.180	15.5	318180†	T/4, W/4	1.31	1.65	1.21	1.44	--	--	--	--
			T/2, W/2	1.37	1.77	1.26	1.61	1.41	1.71	1.37	1.62
				1.38	1.80	1.39	1.63	1.40	1.78	1.36	1.62
								1.24	1.59	1.23	1.58
								1.25	1.63	1.35	1.57
								1.24	1.59	1.23	1.58
								1.25	1.63	1.35	1.57

* T - Thickness, W - Width
† Producer B; all others from Producer A
NOTE: L - Longitudinal; LT - Long-Transverse

TABLE XXI
RATIOS AROUND THE MECHANICAL PROPERTIES AT DIFFERENT LOCATIONS
[AF33(615)-3580]

Alloy and Temper	Section Thickness, in.	Cross-Sectional Area, in. ²	Number	Direction	Location	Tensile Ultimate Stress	Tensile Yield Stress	Compressive Yield Stress	Shear Ultimate Stress	Bearing	
										Ultimate Stress $\sigma/\sigma_{1.5}$	Yield Stress $\sigma/\sigma_{1.5}$
2024-T6510	1.637	2.8	318146	L	Stress-Relieved Stretched Extrusions D/2/W/4	1.01	1.01	1.01	1.00	0.94	0.98
	0.525	1.9	318070**	L	W/2/W/4	0.98	1.02	1.03	0.98	--	--
	0.815	3.9	340418**	L	W/2/W/4	0.96	0.97	0.93	1.00	0.99	1.00
	0.150	5.6	318077	L	W/2/W/4	1.03	0.97	1.00	1.01	0.99	1.01
	1.450	7.2	318021**††	L	W/2/W/4	1.01	0.98	0.99	0.99	0.99	0.98
	1.705	4.8	340213	L	W/2/W/4	1.01	1.00	1.01	0.96	0.98	1.00
	2.520	8.8	318132**	L	W/2/W/4	0.99	1.02	1.00	--	1.00	1.02
	4.000	24.0	340214	L	W/2/W/4	1.02	1.02	1.01	0.98	0.99	1.01
	2.760	29.6	318048	L	W/2/W/4	0.96	1.00	0.99	1.00	0.96	1.04
	4.500	30.7	340288	L	W/2/W/4	1.03	1.06	1.04	1.01	0.99	1.02
				L	W/2/W/4	0.98	0.96	0.95	0.98	1.00	0.98
				L	W/2/W/4	1.00	1.04	1.02	0.96	1.03	1.05
				L	W/2/W/4	--	--	--	--	1.02	1.12
				L	W/2/W/4	0.99	0.98	0.99	0.99	--	--
2024-T6510	0.525	1.9	318024**	L	W/2/W/4	1.00	0.99	1.00	1.00	0.99	1.01
	0.815	3.9	340419**	L	W/2/W/4	1.01	1.01	0.99	1.00	0.99	1.00
	1.150	5.6	318078	L	W/2/W/4	0.96	0.98	1.00	1.00	0.99	1.00
	1.450	7.2	318025**††	L	W/2/W/4	1.00	1.00	0.99	1.00	0.99	1.01

* L - Longitudinal; LT - Long Transverse
† T - Thickness; W - Width; D - Diameter
†† Edge-wise bearing specimens; others - flatwise specimens
** Producer B; all others from Producer A
†† Sample was in the T351 temper
** Sample was in the T651 temper

TABLE IX (Continued)
RATIOS AMONG THE MECHANICAL PROPERTIES AT DIFFERENT LOCATIONS
[AF33(615)-3580]

Alloy and Temper	Section Thickness, in.	Sample Cross-Sectional Area, in. ²	Number	Direction*	Location†	Tensile		Compressive Yield Stress	Shear Ultimate Stress	Bearing	
						Ultimate Stress	Yield Stress			Ultimate Stress $e/D=1.5$	Yield Stress $e/D=2.0$
2024-T3	1.705	4.8	340169	L	TW/12/TW/A	1.00	1.03	1.00	0.98	1.01	1.02
	2.520	8.8	340420**	L	TW/12/TW/A	0.98	1.01	1.00	0.99	0.96	1.00
	2.760	29.6	318079	L	TW/12/TW/A	0.98	0.98	0.99	1.04	0.97	0.98
	4.000	24.0	340225	L	TW/12/TW/A	1.01	1.02	0.99	1.05	1.05	1.05
	4.500	30.7	340389	L	TW/12/TW/A	0.98	0.98	0.98	0.98	0.98	0.98
				L	TW/12/TW/A	0.97	0.98	0.97	0.98	0.96	0.97
6061-T5	1.004	2.0	340423**	L	W/12/A	0.99	0.99	1.01	1.01	1.00	1.02
	1.240	2.7	317907	L	W/12/A	1.01	1.01	0.98	0.99	0.99	0.97
	1.960	4.4	317896	L	W/12/A	0.98	0.98	0.97	0.98	0.98	0.98
	3.000	15.0	340226	L	TW/12/TW/A	0.98	1.02	0.98	0.97	0.98	0.95
	6.500	33.2	317897	L	D/12/A	0.98	0.98	0.99	0.99	0.98	0.98
				L	W/12/A	0.98	0.97	0.97	0.99	0.98	0.98
7075-T6	0.975	7.2	340155	L	W/12/A	0.98	0.97	0.97	0.99	0.99	0.98
	1.188	27.1	317860	L	W/12/A	0.98	0.98	0.97	0.94	0.98	0.98
	2.000	2.1	317861	L	D/12/A	0.98	0.97	0.98	0.96	0.98	0.98
	2.190	17.9	318137**	L	TW/12/TW/A	1.02	0.99	0.95	0.95	1.00	0.98
	2.750	8.2	340404	L	TW/12/TW/A	0.98	0.98	0.98	0.96	0.98	0.98
	3.040	13.8	318138**	L	TW/12/TW/A	0.98	0.98	0.98	0.99	0.99	0.98
				L		0.99	1.00	0.99	0.98	0.99	0.98

* L - Longitudinal; LT - Long Transverse
† T - Thickness; W - Width; D - Diameter
‡ Edge-line bearing specimens; others - Flatness specimens

** Producer B; all others from Producer A

TABLE XXX (Concluded)
RATIOS AMONG THE MECHANICAL PROPERTIES AT DIFFERENT LOCATIONS
AF33(615)-3580

Alloy and Temper	Section Thickness, in.	Cross-Sectional Area, in. ²	Rimmed Direction* Location†	Tensile			Compressive Yield Stress	Shear Ultimate Stress	Bearing	
				Ultimate Stress	Yield Stress	Ratio			Ultimate Stress $\frac{e}{b} \times 2.0$	Yield Stress $\frac{e}{b} \times 2.0$
7075-T6510	3.090	24.3	340391	L LT LT	T4/2/T4/4	0.96	0.95	0.95	0.94	0.96
						0.95	0.95	0.95	0.97*	0.96*
						0.95	0.95	0.95	0.95	0.95*
7075-T6510	0.995	7.2	340292	L	W/2/W/4	0.98	0.98	0.97	0.98	0.98
						0.99	0.99	0.98	0.98	0.98
	1.000	5.7	340439	L	W/2/W/4	0.97	0.96	0.97	0.96	0.97
						0.99	0.98	0.98	0.98	0.98
	2.000	2.1	317948	L	D/2/D/4	0.99	0.98	0.98	0.98	0.98
			340440			0.97	0.97	0.96	0.97	0.96
7075-T6510	0.500	4.2	340424**	L	W/2/W/4	0.96	0.95	0.97	0.96	0.97
						0.99	0.99	0.99	0.99	0.99
						1.02	1.00	0.99	1.00	0.98
7178-T6510	0.625	6.9	317997	L	W/2/W/4	0.98	0.99	0.98	0.98	0.99
						0.98	0.98	0.98	0.98	0.98
	1.200	3.9	318179**	L	W/2/W/4	0.96	0.98	0.98	1.00	0.99*
7075-T6510	2.180	15.5	318140**	L	T4/2/T4/4	0.99	0.99	0.96	0.99	0.97
						0.97	0.95	0.96	0.97	0.95
						1.00	0.98	0.96	0.94	0.95
2024-T42	2.562	6.4	340245	L	W/2/W/4	0.99	0.97	0.97	0.98	0.96
						0.88	0.96	0.95	0.94	0.94
2024-T42	2.562	6.4	340246	L	W/2/W/4	0.95	0.93	0.94	0.94	0.95
						0.94	0.93	0.92	0.94	0.93
6061-T52	1.685	3.9	318091	L	W/2/W/4	1.00	1.01	1.00	1.01	1.02
						1.01	1.02	1.00	1.01	1.02
7075-T6510	1.225	21.2	318098	L	W/2/W/4	0.97	0.96	0.96	0.99	0.98
						0.99	0.98	0.97	1.02	1.01
	2.250	4.1	318100	L	D/2/D/4	0.96	0.98	0.95	0.96	0.97
7075-T6510	1.225	21.2	318099	L	W/2/W/4	0.97	0.96	0.96	0.98	0.99
						0.99	0.98	0.97	1.01	1.02
	2.250	4.1	318101	L	D/2/D/4	0.94	0.93	0.94	0.95	0.96

* L - Longitudinal; LT - Long-Transverse
† W - Width; D - Diameter
* * * * * otherwise bearing specimens; others - Flatwise specimens

** Producer B; all others from Producer A

TABLE XXI

RATIO OF BEARING PROPERTIES IN THE EDGEBISE DIRECTION TO THOSE IN THE PLATISE DIRECTION FOR ALUMINUM ALLOY EXTRUSIONS

Alloy and Temper	Sample			Edgewise/Platise			Alloy and Temper	Sample			Edgewise/Platise			Alloy and Temper
	Section Thick-ness, in.	Gross-Sectional Area, in. ²	Loca-tion*	Dirac-tion*	BUS(2)/BUS(P) e/D=1.5 e/D=2.0	BUS(P)/BUS(P) e/D=1.5 e/D=2.0		Section Thick-ness, in.	Gross-Sectional Area, in. ²	Loca-tion*	Dirac-tion*	BUS(2)/BUS(P) e/D=1.5 e/D=2.0	BUS(P)/BUS(P) e/D=1.5 e/D=2.0	
2024-T3510 **	1.150	5.6	318077	T/2, W/4	L	0.97	0.98	1.188	27.1	317860	T/2, W/4	L	0.86	0.90
	1.200	5.9	317946	T/2, W/4	L	0.97	0.98						0.87	0.90
	1.450	7.3	318021*	T/2, W/4	L	0.97	0.98						0.88	0.90
				T/2, W/2	L	0.97	0.98	2.190	17.0	318137*	T/4, W/4	L	0.90	0.96
	1.705	4.8	340213	T/2, W/2	L	0.97	0.98						0.88	0.90
	2.520	6.8	318133*	T/2, W/2	L	0.97	0.98						0.88	0.90
				T/2, W/2	L	0.97	0.98						0.88	0.90
	4.000	24.0	340214	T/2, W/4	L	0.97	0.98	2.750	8.2	340404	T/4, W/4	L	0.85	0.90
				T/2, W/2	L	0.97	0.98						0.85	0.90
	2.760	29.1	318048	T/2, W/2	L	0.97	0.98	3.040	12.8	318138*	T/4, W/4	L	0.95	0.99
2024-T6510 **	1.150	5.6	318078	T/2, W/4	L	0.97	0.98						0.95	0.99
	1.200	5.9	317895	T/2, W/4	L	0.97	0.98						0.95	0.99
	1.450	7.3	318025*	T/2, W/4	L	0.97	0.98	2.750	8.2	340440	T/4, W/4	L	0.95	0.99
				T/2, W/2	L	0.97	0.98						0.95	0.99
	1.705	4.8	340160	T/2, W/2	L	0.97	0.98						0.95	0.99
	2.520	8.8	340420*	T/2, W/2	L	0.97	0.98						0.95	0.99
				T/2, W/2	L	0.97	0.98						0.95	0.99
	2.760	29.1	318079	T/2, W/4	L	0.97	0.98	1.200	3.9	318139*	T/4, W/4	L	0.97	0.99
				T/2, W/2	L	0.97	0.98						0.97	0.99
	4.000	24.0	340225	T/2, W/2	L	0.97	0.98	2.180	15.5	318140*	T/4, W/4	L	0.89	0.94
6061-T6510	1.150	5.6	318078	T/2, W/4	L	0.97	0.98						0.97	0.99
	1.200	5.9	317895	T/2, W/4	L	0.97	0.98						0.97	0.99
	1.450	7.3	318025*	T/2, W/4	L	0.97	0.98						0.97	0.99
				T/2, W/2	L	0.97	0.98						0.97	0.99
	1.705	4.8	340160	T/2, W/2	L	0.97	0.98						0.97	0.99
	2.520	8.8	340420*	T/2, W/2	L	0.97	0.98						0.97	0.99
				T/2, W/2	L	0.97	0.98						0.97	0.99
	2.760	29.1	318079	T/2, W/4	L	0.97	0.98						0.97	0.99
				T/2, W/2	L	0.97	0.98						0.97	0.99
	4.000	24.0	340225	T/2, W/2	L	0.97	0.98						0.97	0.99
Extrusions in the Heat-Treated-by-User Temper														
6061-T6510	1.150	5.6	318078	T/2, W/4	L	0.97	0.98	2.562	6.4	340245	T/4, W/4	L	0.98	0.99
	1.200	5.9	317895	T/2, W/4	L	0.97	0.98						0.98	0.99
	1.450	7.3	318025*	T/2, W/4	L	0.97	0.98						0.98	0.99
				T/2, W/2	L	0.97	0.98						0.98	0.99
	1.705	4.8	340160	T/2, W/2	L	0.97	0.98						0.98	0.99
	2.520	8.8	340420*	T/2, W/2	L	0.97	0.98						0.98	0.99
				T/2, W/2	L	0.97	0.98						0.98	0.99
	2.760	29.1	318079	T/2, W/4	L	0.97	0.98						0.98	0.99
				T/2, W/2	L	0.97	0.98						0.98	0.99
	4.000	24.0	340225	T/2, W/2	L	0.97	0.98						0.98	0.99
Extrusions in the Heat-Treated-by-User Temper														
6061-T6510	1.150	5.6	318078	T/2, W/4	L	0.97	0.98	2.562	6.4	340246	T/4, W/4	L	0.98	0.99
	1.200	5.9	317895	T/2, W/4	L	0.97	0.98						0.98	0.99
	1.450	7.3	318025*	T/2, W/4	L	0.97	0.98						0.98	0.99
				T/2, W/2	L	0.97	0.98						0.98	0.99
	1.705	4.8	340160	T/2, W/2	L	0.97	0.98						0.98	0.99
	2.520	8.8	340420*	T/2, W/2	L	0.97	0.98						0.98	0.99
				T/2, W/2	L	0.97	0.98						0.98	0.99
	2.760	29.1	318079	T/2, W/4	L	0.97	0.98						0.98	0.99
				T/2, W/2	L	0.97	0.98						0.98	0.99
	4.000	24.0	340225	T/2, W/2	L	0.97	0.98						0.98	0.99

** Bearing specimen failed before reaching yield stress

†† Sample was in the T7511 temper

** Sample was in the T6511 temper

* T = Thickness; W = Width; D = Diameter

* L = Longitudinal; LT = Long-Transverse

* Producer B: all others from Producer A

TABLE XVII

RESULTS OF PLASTICITY-REDUCED STRESS OF SIMILAR-ROD-SIZED STRUCTURES
OF STRESS-RELIEVED STRENGTHENED ALUMINUM ALLOY EXTENSIONS

(MPS(615)-398)

Sample	Sectional Thickness, in.	Direction Number	Specimen	At Initial Indication of Crack Growth				At 5 Per Cent Seamed Offset			
				Crack Length, in.	Width of Crack, in.	Ratio of Load to Initial Load	Ratio of Load to Initial Load	Crack Length, in.	Width of Crack, in.	Ratio of Load to Initial Load	Ratio of Load to Initial Load
0.271	0.40	317994	12	1.502	0.271	0.52	0.52	1.48	0.271	0.52	0.52
0.295	2.8	317992	12	1.502	0.271	0.52	0.52	1.48	0.271	0.52	0.52
0.310	10.1	317996	12	1.502	0.271	0.52	0.52	1.48	0.271	0.52	0.52
0.642	5.8	317995	12	1.502	0.271	0.52	0.52	1.48	0.271	0.52	0.52
1.490	7.3	3180214	12	1.502	0.271	0.52	0.52	1.48	0.271	0.52	0.52
4.000	24.0	340214	12	1.502	0.271	0.52	0.52	1.48	0.271	0.52	0.52
2.760	20.6	318048	12	1.502	0.271	0.52	0.52	1.48	0.271	0.52	0.52

1. Longitudinal; 2. Transverse;
3. After fatigue crack;
4. After Pop-in; 5. After maximum load without significant prior deviation from linearity;
6. Initial significant deviation from linearity;
7. Maximum load

Combined $(P/A + K_0/L)$ stress at tip of crack
 $K_0 = \frac{P}{A} \left[1 + \frac{1}{2} \left(\frac{L}{r} \right)^2 \right]$; ref. ASTM 979-611.

K_0 = Coefficient value of plane-stress stress-intensity factor, psi V^{1/2}.
 P = Load, lb., at unstable crack growth
 A = Cross section, in.
 L = Crack length, in.

$K_0 = K_0 + K_0' =$ actual crack length, in., plus plastic zone correction factor, in.
 $K_0' =$ plastic zone correction factor, in.
 $K_0 =$ tensile yield stress, psi
 $K_0' =$ plastic zone correction factor, in.

Product 8; all others are Product 4

TABLE XXII (Continued)
RESULTS OF FRACTURE-TOUGHNESS TESTS OF SINGLE-TENSILE-SWITCHED SPECIMENS
OF STRESS-RELIEVED STRETCHED ALUMINUM ALLOY EXTENSIONS
[A773(6:5)-2580]

Sample	Sectional Thickness, in.	Direction and Fracture (See Fig. 7)	Specimen		At Initial Indication of Crack Growth					At 5 Per Cent Secant Offset				
			Width (in.)	Thickness (in.)	Crack Length (in.)	Yield Strength (psi)	Ultimate Tensile Strength (psi)	Modulus (psi)	Reduction of Area (%)	Yield Strength (psi)	Ultimate Tensile Strength (psi)	Modulus (psi)	Reduction of Area (%)	Reduction of Area (%)
6.346	55.2	317697	6.000	1.007	2.17	1.007	1.007	1.007	0.57	1.007	1.007	1.007	0.57	0.57
			6.000	1.007	2.17	1.007	1.007	1.007	0.57	1.007	1.007	1.007	0.57	0.57
			6.000	1.007	2.17	1.007	1.007	1.007	0.57	1.007	1.007	1.007	0.57	0.57
0.438	7.2	317699	3.000	0.500	1.07	0.500	0.500	0.500	0.11	0.500	0.500	0.500	0.11	0.11
			3.000	0.500	1.07	0.500	0.500	0.500	0.11	0.500	0.500	0.500	0.11	0.11
			3.000	0.500	1.07	0.500	0.500	0.500	0.11	0.500	0.500	0.500	0.11	0.11
1.168	27.1	317660	6.000	1.120	2.18	1.120	1.120	1.120	0.10	1.120	1.120	1.120	0.10	0.10
			6.000	1.120	2.18	1.120	1.120	1.120	0.10	1.120	1.120	1.120	0.10	0.10
			6.000	1.120	2.18	1.120	1.120	1.120	0.10	1.120	1.120	1.120	0.10	0.10
2.190	17.0	318176*	3.000	1.010	1.00	1.010	1.010	1.010	0.11	1.010	1.010	1.010	0.11	0.11
			3.000	1.010	1.00	1.010	1.010	1.010	0.11	1.010	1.010	1.010	0.11	0.11
			3.000	1.010	1.00	1.010	1.010	1.010	0.11	1.010	1.010	1.010	0.11	0.11
3.520	13.2	318196*	4.500	1.446	1.51	1.446	1.446	1.446	0.11	1.446	1.446	1.446	0.11	0.11
			4.500	1.446	1.51	1.446	1.446	1.446	0.11	1.446	1.446	1.446	0.11	0.11
			4.500	1.446	1.51	1.446	1.446	1.446	0.11	1.446	1.446	1.446	0.11	0.11
6.482	7.2	317642	6.000	0.795	1.01	0.795	0.795	0.795	0.09	0.795	0.795	0.795	0.09	0.09
			6.000	0.795	1.01	0.795	0.795	0.795	0.09	0.795	0.795	0.795	0.09	0.09
			6.000	0.795	1.01	0.795	0.795	0.795	0.09	0.795	0.795	0.795	0.09	0.09
6.614	6.3	317607	6.000	0.640	0.64	0.640	0.640	0.640	0.04	0.640	0.640	0.640	0.04	0.04
			6.000	0.640	0.64	0.640	0.640	0.640	0.04	0.640	0.640	0.640	0.04	0.04
			6.000	0.640	0.64	0.640	0.640	0.640	0.04	0.640	0.640	0.640	0.04	0.04
7.136	10.5	318140*	6.000	0.909	0.91	0.909	0.909	0.909	0.09	0.909	0.909	0.909	0.09	0.09
			6.000	0.909	0.91	0.909	0.909	0.909	0.09	0.909	0.909	0.909	0.09	0.09
			6.000	0.909	0.91	0.909	0.909	0.909	0.09	0.909	0.909	0.909	0.09	0.09

* - Direction of crack growth is transverse.
* - Direction of crack growth is longitudinal.
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* - Direction of crack growth is diagonal.

TABLE XXIII

RESULTS OF FRACTURE-TOUGHNESS TESTS OF SINGLE-LINE-NOTCHED SPECIMENS OF ALUMINUM ALLOY ELIMINATIONS IN THE "HEAT-TREATED-AT-TEMPERATURE"

[ASTM (613)-2580]

Specimen	Sectional Area, in. ²	Direction and Number (See Fig. 7)	Specimen			At Initial Indication of Crack Growth				At 5 Pct. Cent. Seemant Offset			
			Width (in.)	Thickness (in.)	Crack Length (in.)	Nature of Pop-In	Load (lb.)	Gross Stress (psi)	Net Stress (psi)	Load (lb.)	Gross Stress (psi)	Net Stress (psi)	$\frac{K_{IC}}{B\sqrt{a}}$, in./in. ^{3/2}
C-300	6.3	310084	11	0.975	0.295	0.15	3 750	13 700	22 600	4 100	14 800	29 400	0.15
		72	1	0.975	0.295	0.15	3 900	14 100	24 700	4 100	14 700	29 400	0.15
		72	1	0.975	0.295	0.15	3 900	14 100	24 700	4 100	14 700	29 400	0.15
1.225	21.2	318098a	11	4.500	1.150	1.41	28 400	5 500	18 800	28 400	5 500	18 800	0.91
		72	2	4.500	1.150	1.41	28 400	5 500	18 800	28 400	5 500	18 800	0.91
		72	2	4.500	1.150	1.41	28 400	5 500	18 800	28 400	5 500	18 800	0.91
1.225	21.2	318098a	11	4.500	1.150	1.61	28 400	5 500	18 800	28 400	5 500	18 800	0.91
		72	2	4.500	1.150	1.61	28 400	5 500	18 800	28 400	5 500	18 800	0.91
		72	2	4.500	1.150	1.61	28 400	5 500	18 800	28 400	5 500	18 800	0.91

L = Longitudinal, T = Transverse

At 5 pct. of fracture-toughness test, i.e., after fatigue cracking

P = Clear Pop-In, below maximum load; without significant prior deviation from linearity

P = Initial significant deviation from linearity

P/A = Load (P/A + Mc/A) stress at tip of crack

K_{IC} = $\frac{P}{B\sqrt{a}}$ [1.99 - 32 (B)² + 117 (B)³]; ref: ASTM SPT 411K_{IC} = Candidate value of plane-strain stress-intensity factor, psi/in.

B = Load, lb.; a = unstable crack growth

B = Thickness, in.

B = Gross width, in.

a = a₀ + $\frac{K_{IC}^2}{\sigma_y^2}$ = actual crack length, in., plus plastic-zone correction factor, in.σ_y = tensile yield stress, psi

P = Product B; all stresses are Product A

TABLE XXIV

RESISTANCE TO STRESS-CORROSION CRACKING OF STRESS-RELIEVED
STRETCHED ALUMINUM ALLOY EXTRUSIONS

Alloy	Section Thickness, in.	Number	Exposure: 3.5% NaCl Solution by Alternate Immersion Stretched - 75% Yield Strength				Short Transverse			
			Longitudinal		Per cent Loss in Tensile Strength††		Long Transverse		Per cent Loss in Tensile Strength††	
			P/N†	Days†	Strength††	Days†	P/N†	Days†	Strength††	Days†
2024-T6510	0.250	340154	0/2	75	--		2/2	2,2***	--	--
	0.625	340486								
	1.755	340487								
2024-T7510	0.255	317042	0/2	84	29		0/2	84	22	--
	0.510	317043	0/2	84	34		0/2	84	25	--
	0.970	317044	0/2	84	18		1/2	31 (OK 84)	42	6.6
	1.200	317045	0/2	84	14		2/2	17, 7	--	6.6
	2.760	318048	0/2	84	25		2/2	12, 26	--	40 (OK 84)
	4.000	340214**	0/2	75			2/2	2, 5	--	84, 84§
2024-T8510	0.255	317890	0/2	84	5		0/2	84	8	--
	0.510	317892	0/2	84	6		0/2	84	7	--
	0.970	317893	0/2	84	6		0/2	84	9	84
	1.200	317895	0/2	84	6		0/2	84	10	84
	2.760	318079	0/2	84	9		0/2	84	17	84
	4.000	340225**	0/2	75			0/2	75	0/2	84
6061-T6510	0.215	317057	0/2	84	4		0/2	84	0	--
	0.430	317058	0/2	84	0		0/2	84	0	--
	1.200	317059	0/2	84	0		0/2	84	0	--
	1.940	317065	0/2	84	0		0/2	84	0	--
	3.000	340227	0/2	75			0/2	75	0/2	84
7075-T6510	0.215	317054	0/2	84	3		1/2	17 (OK 84)	7	--
	0.430	317055	0/2	84	3		0/2	84	--	--
	0.970	340155*	0/2	28	2		0/2	28	--	--
	1.188	317860*	0/2	84	2		0/2	84	2	6.6
	2.160	318127	0/2	84	7		1/2	8 (OK 84)	18	4.4
	3.040	318133	0/2	84	5		2/2	4, 4	--	9.0
7075-T73510	0.275	317900	0/2	84	0		0/2	84	0	--
	0.430	317910	0/2	84	2		0/2	84	3	--
7075-T73510	0.975	340232*	0/2	28			0/2	28	--	--
	3.095	340392	0/2						--	--
7075-T6510	0.251	340253	0/2	75			0/2	28	--	--
7178-T6510	0.625	317097	0/2	84	9		1/2	81 (OK 84)	14	--
	1.200	318139*	0/2	84	10		2/2	7, 7	--	4.5
	2.190	318140	0/2	84	11		2/2	7, 7	--	4.5

* P/N denotes number of specimen failed over number exposed

†† Tests in progress for periods shown, with maximum duration of 84 days

††† Results are average values for tensile tests of specimens which did not fail by stress-corrosion cracking.

* Short transverse yield strengths determined by tests of duplicate 0.050" diameter tension specimens

** The directionality of these sections is being determined metallographically.

*** Failure occurred outside the reduced section beneath the protective coating used to isolate all parts of the stressing frame.

§ Accumulated corrosion products prevented detection of these failures until specimens were chemically cleaned at termination of the exposure period.

TABLE XXIV
RESISTANCE TO STRESS-CORROSION CRACKING OF ALUMINUM ALLOY
EXTRUSIONS IN THE "HEAT-TREATED-BY-USER" TEMPER

[AF33(615)-3580]

Alloy	Section Thickness, in.	Number	Exposure: 3.5% NaCl Solution by Alternate Immersion Stressing - 75% Yield Strength									
			Longitudinal					Short Transverse				
			P/N†	Days††	Per cent Loss in Tensile Strength†††	P/N†	Days††	Per cent Loss in Tensile Strength†††	P/N†	Days††	Per cent Loss in Tensile Strength†††	Days††
2014-T62	0.300	318084	1/2	24§(OK 84)	11	1/2	24§(OK 84)	13	--	--	--	--
2024-T42	0.430	340241	0/2	28		0/2	28		--	2/2	46, 46§§	
	2.562	340245	0/2	28		0/2	28	2, 2	--	2/2		
2024-T62	0.430	340242	0/2	28		0/2	28		--	0/2	46	
	2.562	340246	0/2	28		0/2	28		--	0/2		
6061-T62	0.246	318090	0/2	84	0	0/2	84		--	0/2	84	
	1.625	318091**	0/2	84	2	0/2	84		--	0/2		
7075-T6	0.350	318096	0/2	84	6	0/2	84	7, 6	--	2/2	7, 7	
	1.225	318098**	0/2	84	4	0/2	84		--	2/2		
7075-T73X	0.350	318097	0/2	84	2	0/2	84	4	--	0/2	84	
	1.225	318098**	0/2	84	1	0/2	84	5	--	0/2		
7178-T6	0.403	340249	0/2	28		0/2	28		--	--	--	--

** Short transverse yield strengths determined by tests of duplicate 0.050" diameter tension specimens

† P/N denotes number of specimens failed over number exposed.

†† Tests in progress for periods shown, with maximum duration of 84 days

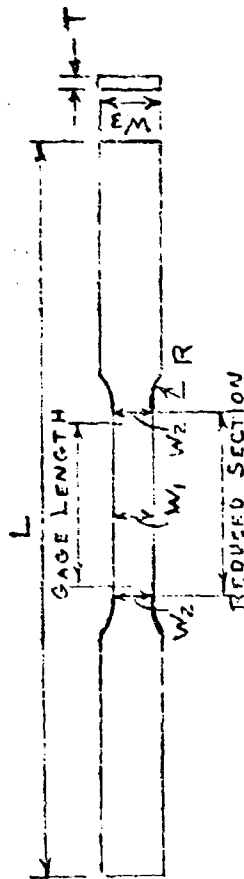
††† Results are average values for tension tests of specimens which did not fail by stress-corrosion cracking.

... Short transverse yield strengths determined by tests of duplicate 0.050" diameter tension specimens

§ Failure occurred outside the reduced section beneath the protective coating used to isolate all parts of the stressing frame.

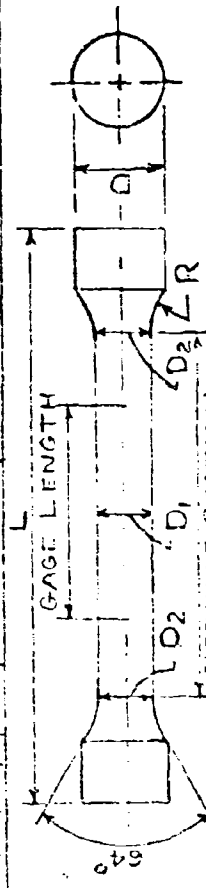
§§ Accumulated corrosion products prevented ready detection of these failures. Specimens were chemically cleaned to confirm suspected failure.

SHEET-TYPE
SPECIMENS

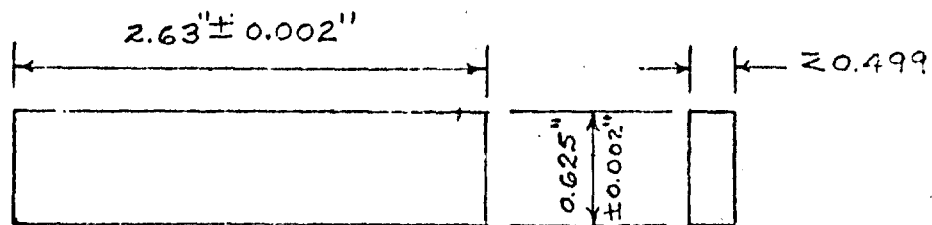


WIDTH, IN.			GAGE LENGTH, IN.	REDUCED SECTION LENGTH, IN.	RADIUS (R), IN.	THICKNESS (T), IN.	LENGTH (L), IN.
W1	W2	W3					
0.500 ± 0.010	W1 + 0.005 0.505	3/4	2.000 ± 0.002	2 - 1/4	7/8	≤ 0.499	9 MIN.
0.250 ± 0.002	W1 + 0.003 0.253	3/8	1.000 ± 0.002	1 - 1/4	3/8	≤ 0.250	4 MIN.
0.125 ± 0.001	W1 + 0.002 0.127	3/16	0.500 ± 0.002	5/8	3/16	≤ 0.125	2 - 1/4 MIN.

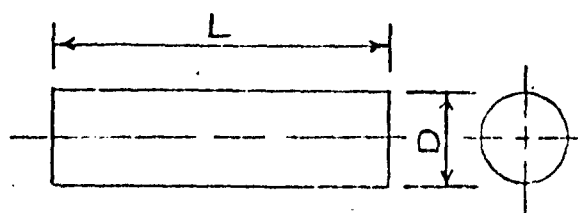
TAPERED-SEAT
SPECIMENS



DIAMETER, IN.		GAGE LENGTH, IN.	REDUCED SECTION LENGTH, IN.	RADIUS (R), IN.	DIAMETER (D), IN.	LENGTH (L), IN.
D1	D2					
0.500 ± 0.005	D1 + 0.005 0.505	2.000 ± 0.002	3 - 1/8	3/8	3/4	4 - 3/4
0.357 ± 0.004	D1 + 0.003 0.360	1.400 ± 0.002	2 - 15/64	17/64	17/32	3 - 3/8
0.250 ± 0.003	D1 + 0.002 0.252	1.000 ± 0.002	1 - 9/16	3/16	3/8	2 - 3/8
0.160 ± 0.002	D1 + 0.001 0.161	0.640 ± 0.002	1	0.120	15/64	1 - 1/2
0.125 ± 0.001	D1 + 0.001 0.126	0.500 ± 0.002	25/32	3/32	3/16	1 - 1/4

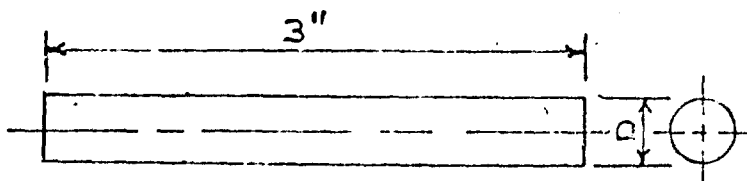


Sheet-Type Compressive Specimen



NOMINAL DIAM, IN.	D, IN.	L, IN.
1/2	$\frac{0.4980}{0.4950}$	$\frac{1-29/32}{1-27/32}$
7/16	$\frac{0.4390}{0.4360}$	$\frac{1-21/32}{1-5/8}$
3/8	$\frac{0.3765}{0.3735}$	$\frac{1-17/32}{1-1/2}$

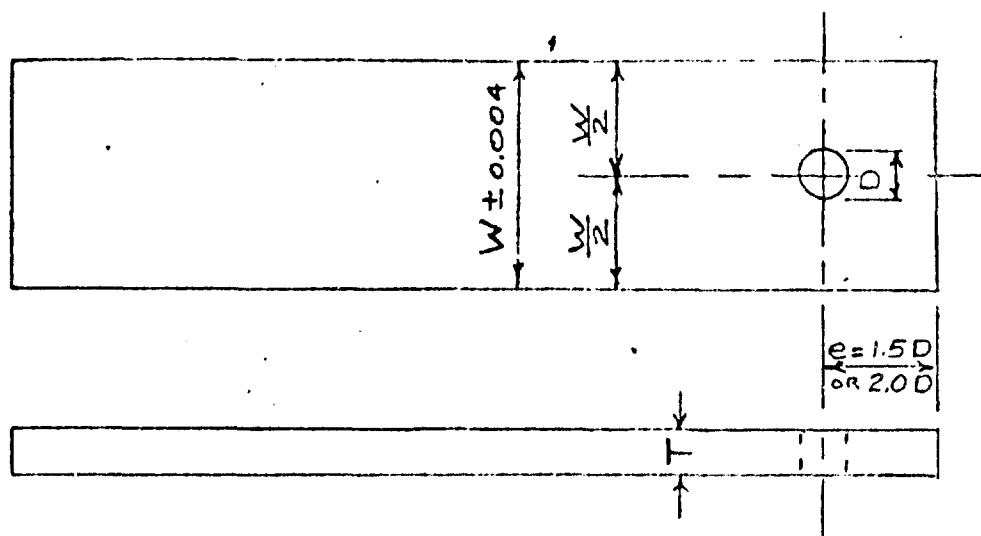
Round Compressive Specimen



NOMINAL DIAM, IN.	D, IN.
3/8	$\frac{0.3730}{0.3720}$
1/4	$\frac{0.2490}{0.2480}$
3/16	$\frac{0.1865}{0.1855}$

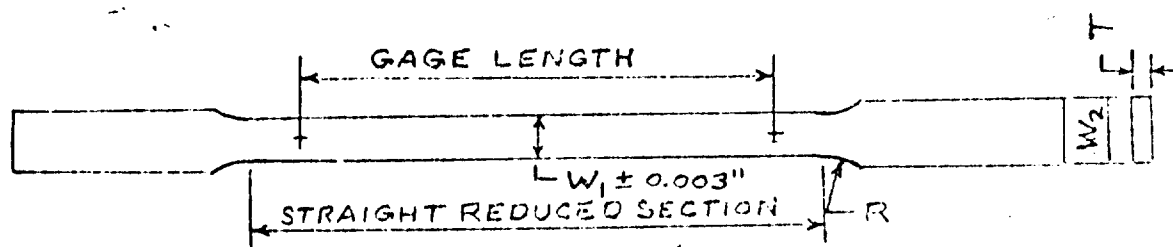
Shear Specimen

Fig. 2. General Dimensions of Compressive and Shear Specimens



TYPE	T, IN.	W, IN.	D, IN.
A	0.063	1	$\frac{0.2500}{0.2505}$
B	0.040-0.074	1-1/2	$\frac{0.2500}{0.2505}$
C	0.075-0.109	1-1/2	$\frac{0.3750}{0.3755}$
D	0.110-0.250	2	$\frac{0.5000}{0.5005}$

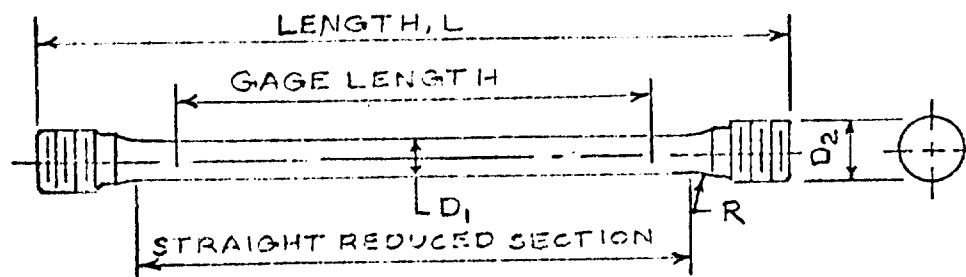
Fig. 3. General Dimensions of Bearing Specimens



WIDTH, IN.		GAGE LENGTH, IN.	REDUCED SECTION LENGTH, IN.	RADIUS (R), IN.	THICKNESS (T), IN.
W_1	W_2				
0.500 ± 0.003	$3/4$	6.000 ± 0.002	7*	$7/8$	≥ 0.499
0.250 ± 0.002	$3/8$	1.000 ± 0.002	$1-1/2$	$3/8$	≥ 0.250

* FOR SOME LONG-TRANSVERSE SPECIMENS, GAGE LENGTHS - 4 IN.
REDUCED-SECTION LENGTHS - 5 IN.

Sheet-Type Specimens

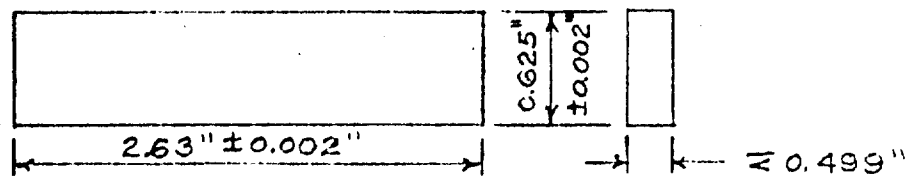


DIAMETER, IN.		GAGE LENGTH, IN.	REDUCED SECTION LENGTH, IN.	RADIUS (R), IN.	LENGTH (L), IN.
D_1	D_2				
0.500 ± 0.003	$3/4$	6.000 ± 0.002	7	$5/8$	$9-1/2$
0.500 ± 0.003	$3/4$	4.000 ± 0.002	5	$5/8$	$7-1/2$
0.500 ± 0.003	$3/4$	2.000 ± 0.002	3†	$5/8$	$5-1/2$ †
0.438 ± 0.003	$5/8$	2.000 ± 0.002	$2-7/8$ †	$\geq D_1$	$5-1/4$ †
0.375 ± 0.002	$9/16$	2.000 ± 0.002	$2-3/4$	$\geq D_1$	5

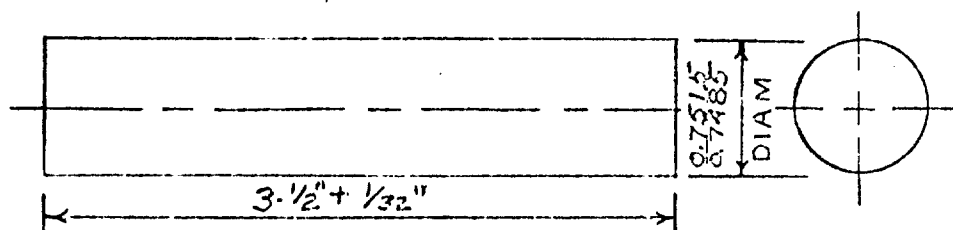
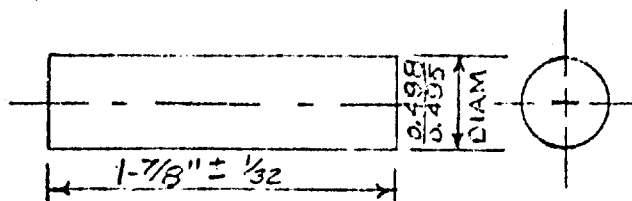
† FOR SHORTER LONG-TRANSVERSE SPECIMENS, GAGE LENGTHS - 1 IN.,
REDUCED-SECTION LENGTHS - 1 IN. PLUS TWO TIMES D_1

Round Specimens

Fig. 4 General Dimensions of Tensile Specimens For
Modulus and Stress-Strain Tests



Sheet-Type Specimen



Round Specimens

Fig. 5 General Dimensions of Compressive Specimens
For Modulus and Stress-Strain Tests

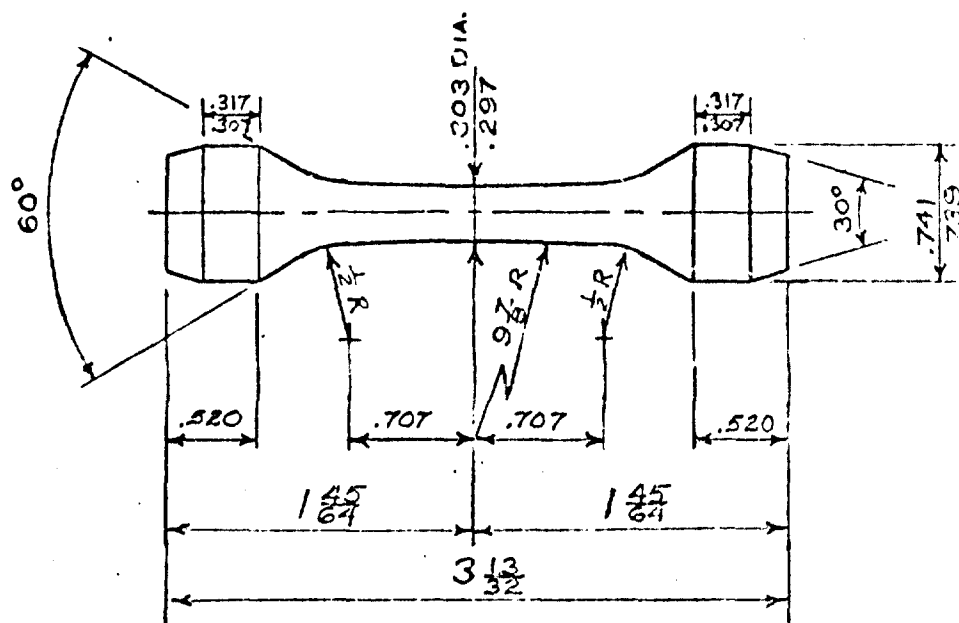


Fig. 6 Axial-Stress Fatigue Specimen

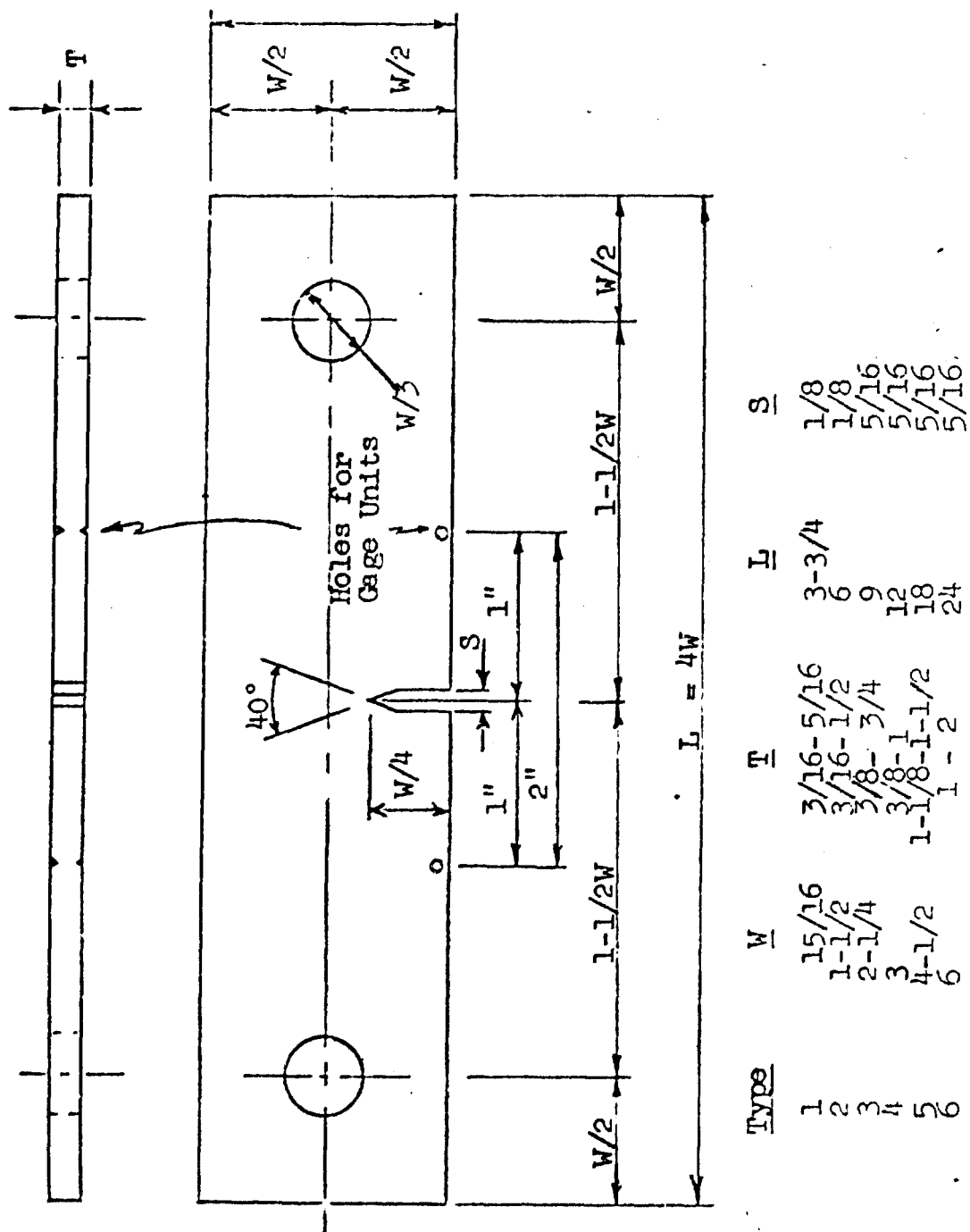
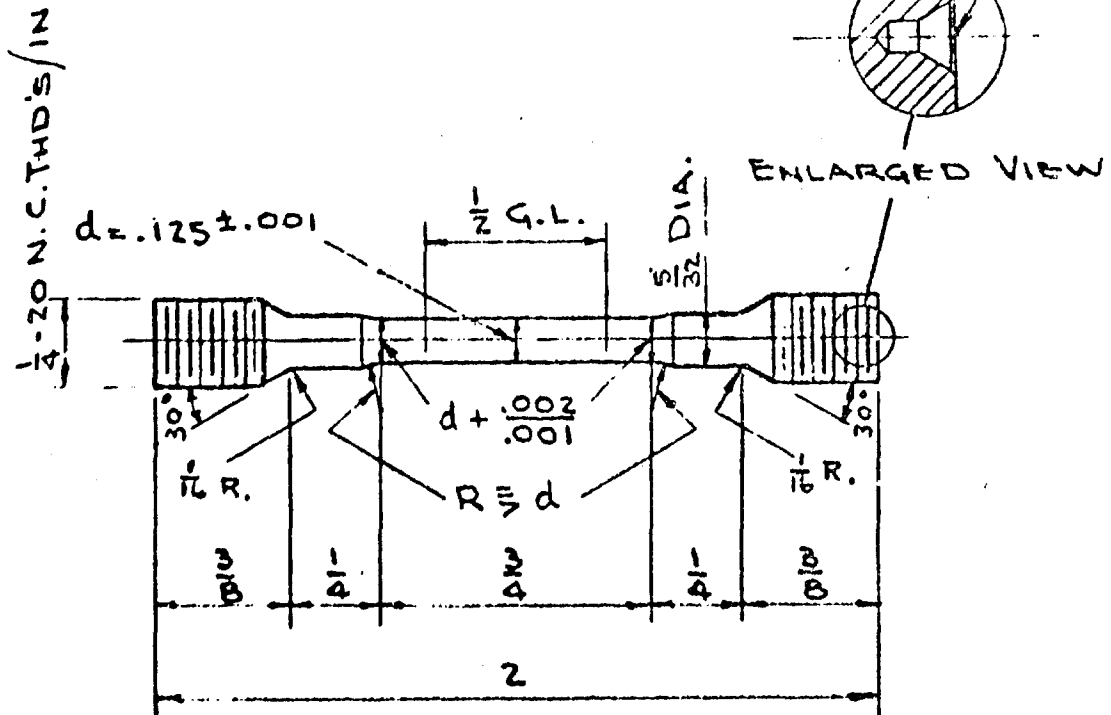


Fig. 7 Single-Edge-Notched Fracture-Toughness Specimens.

L-7034-RK

#731 SIZE 11
BELL TYPE
CENTER DRILL
BOTH ENDS



SPECIMEN  REQ'D

MACHINING SETUP	D-6996-RK
TOOL BIT	D-7647-RK
CENTER & FACE TOOL	D-7444-RK
TEMPLATE	D-6989-RK
DRIVER	D-6981-RK

ALUMINUM COMPANY OF AMERICA
ALCOA RESEARCH LABORATORIES
MECHANICAL ENGINEERING DIVISION NEW KENSINGTON, PA.

MECHANICAL TESTING DIVISION

1/8-DIA. THREADED END TENSILE

SPECIMEN — DETAILS

DATE	NO.	REVISION RECORD	DR	CK
1-15-58		SUPERSEDES D-2276-RK		

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SCALE 24=1'-0"	DATE 1-15-58
L-7034-RK	DRAWN R. MCCLELLAN
	CHECK <i>R. McClellan</i>
	APPR <i>R. McClellan</i>

PR 3412-1208
Fig. 8

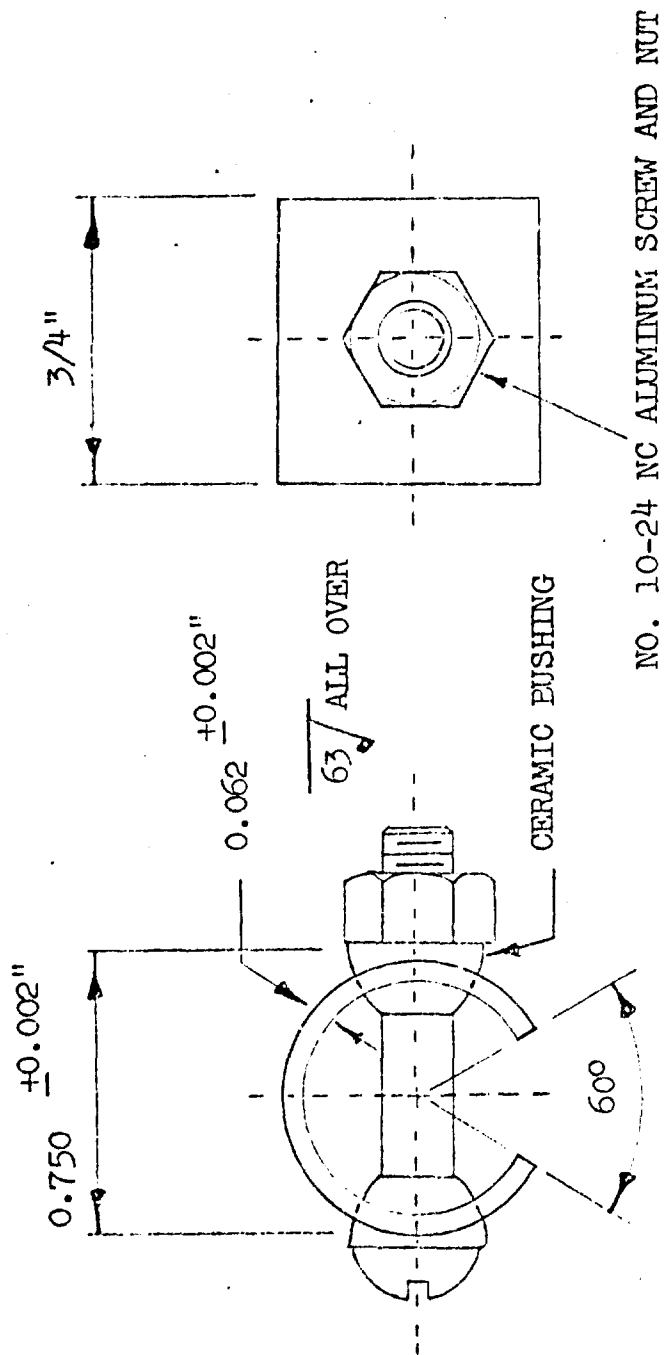


FIGURE 9. C-RING ASSEMBLY FOR PRESS-CORROSION TESTS

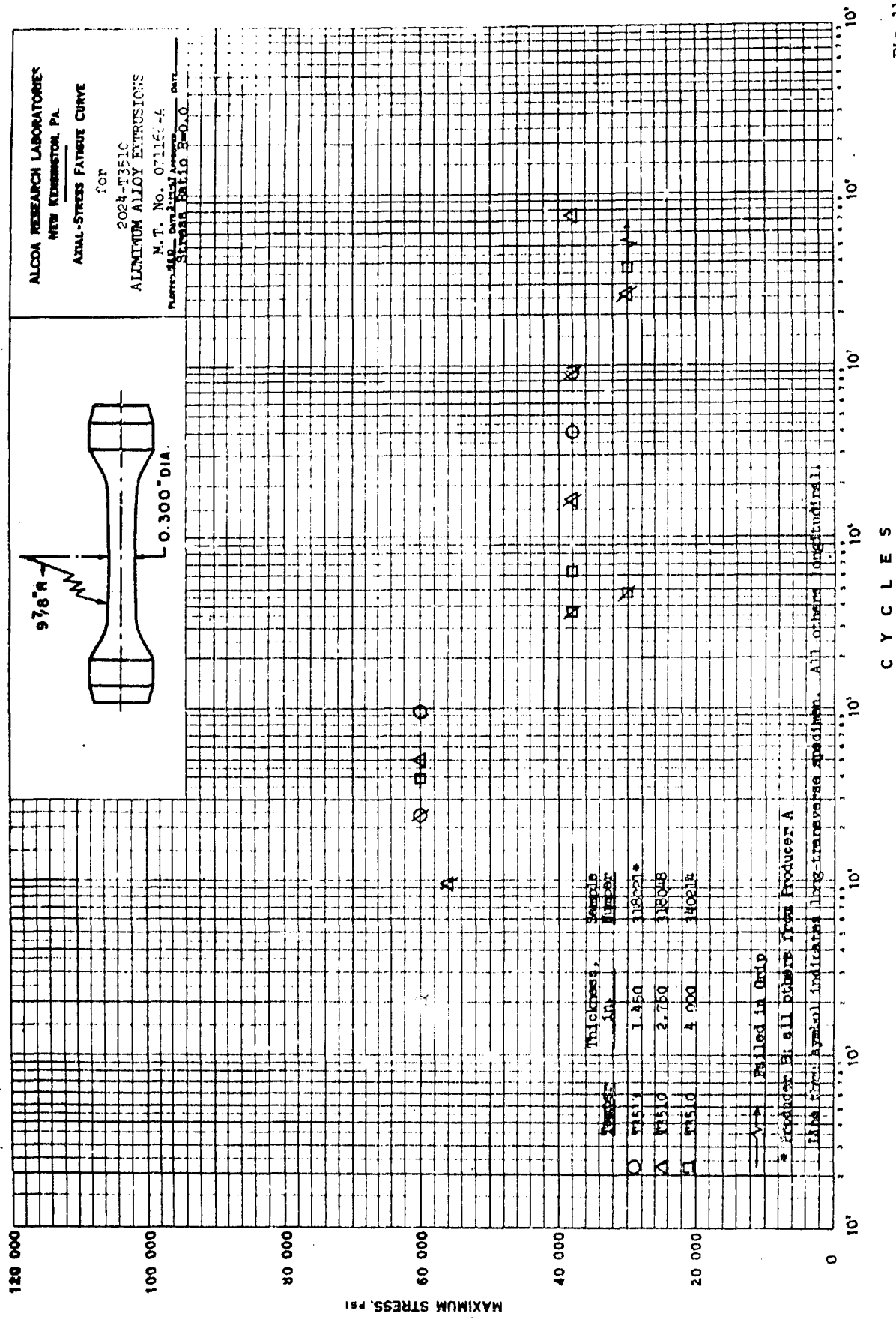


Fig. 11

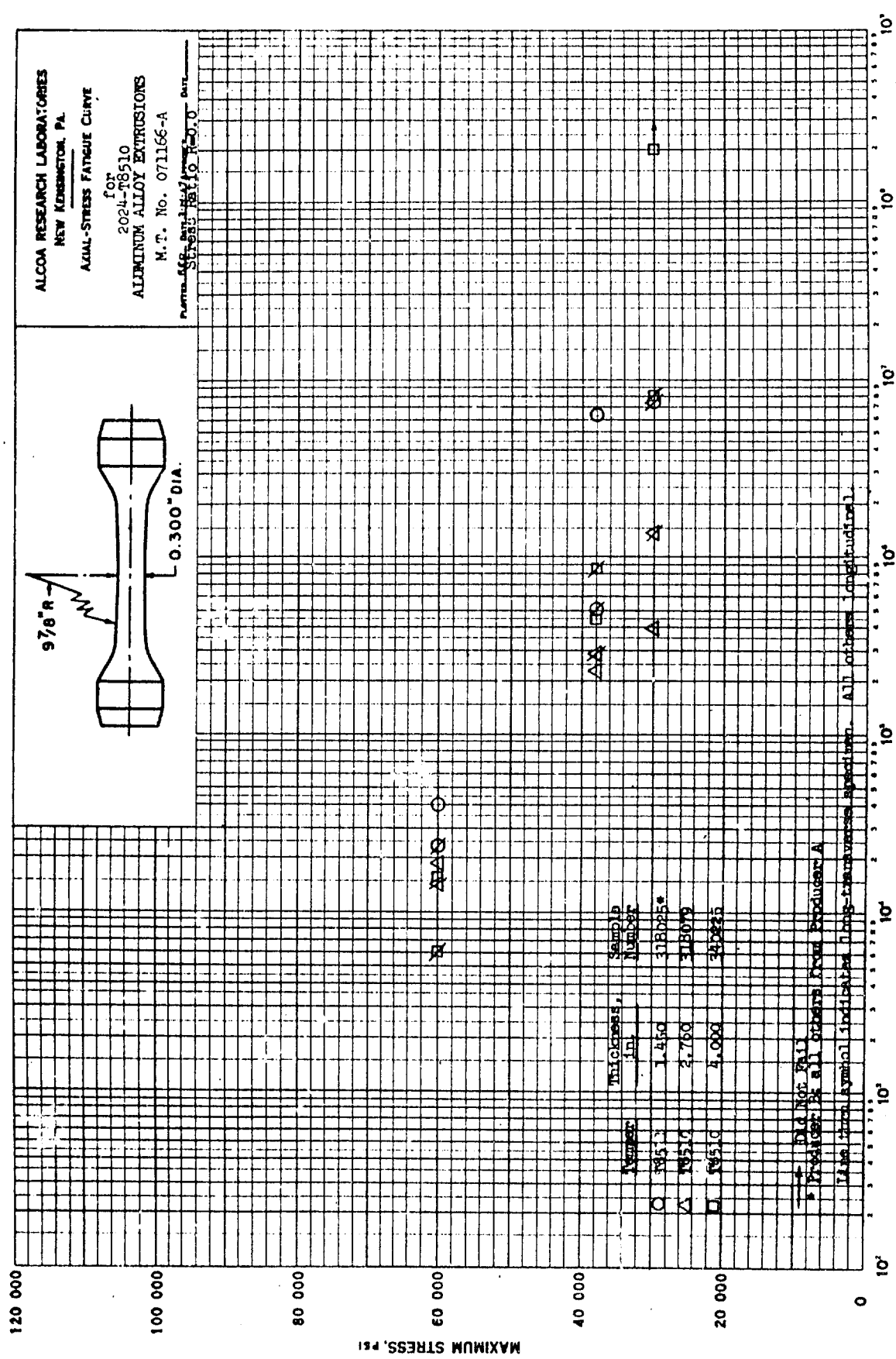


FIG. 12

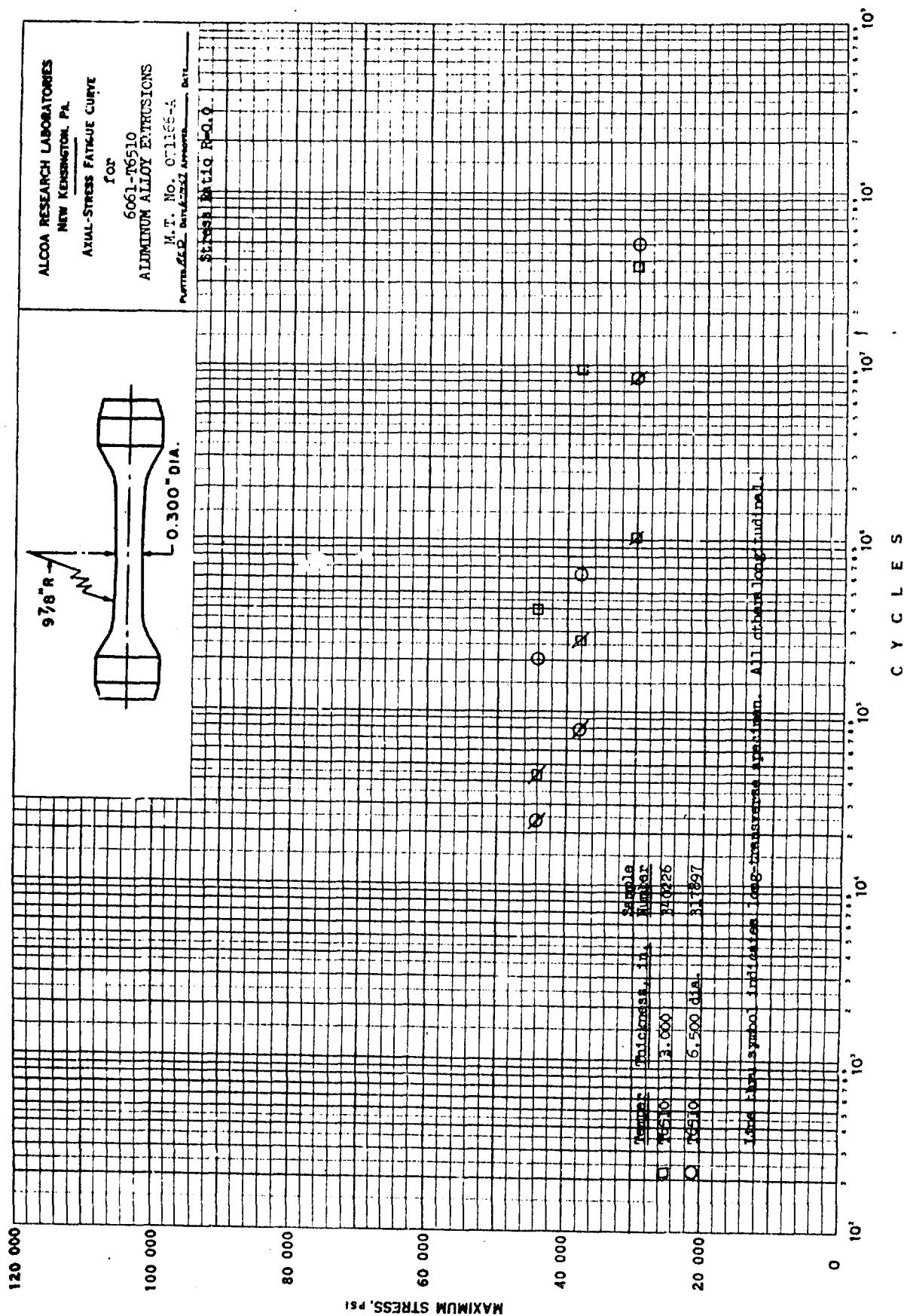


Fig. 13

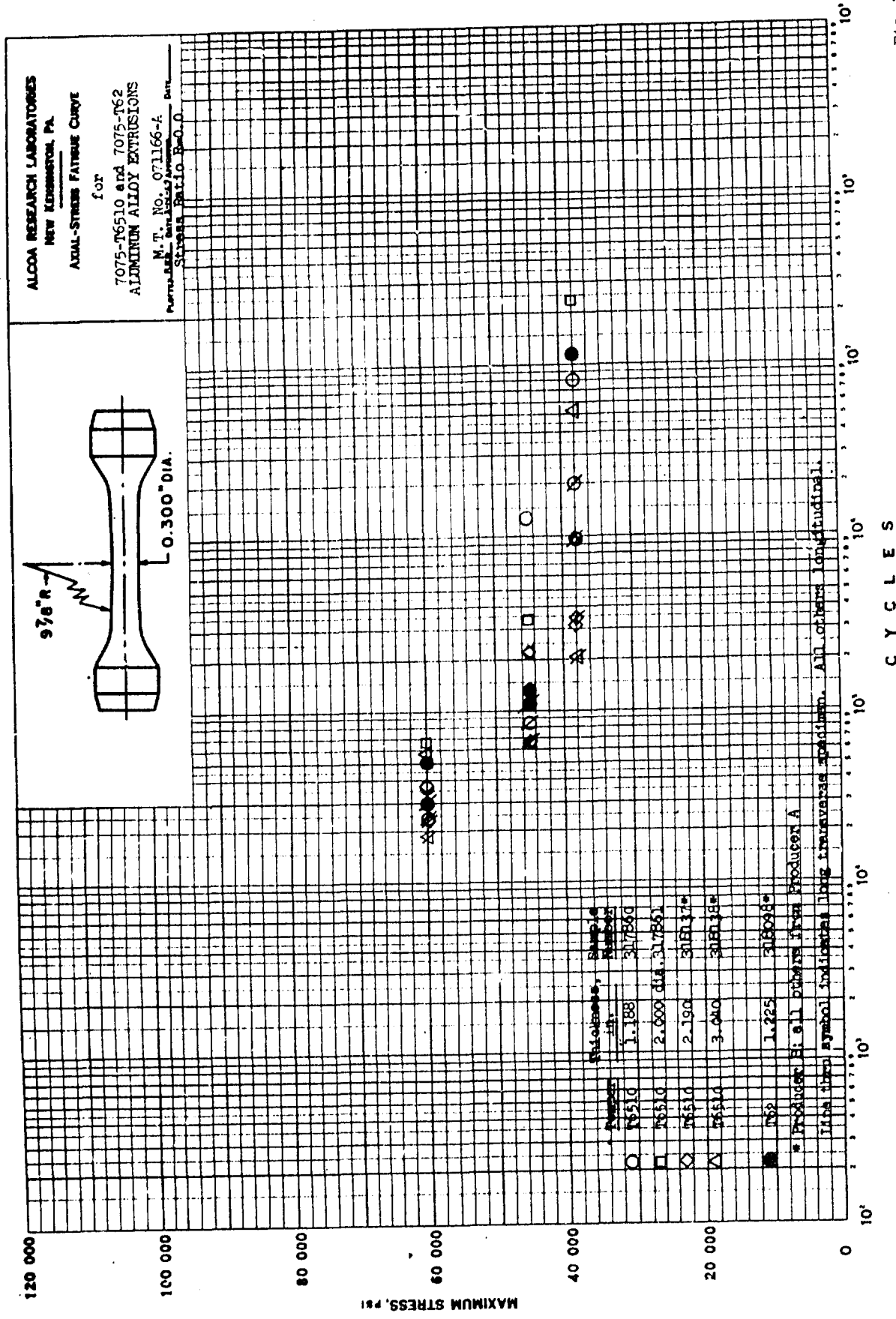


FIG. 14

